

Beyond Probability

A pragmatic approach to uncertainty quantification in engineering

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Wishful thinking

- Using inputs or models because they are convenient, or because you hope they're true

Kansai International Airport



- 30 km from Kobe in Osaka Bay
- Artificial island made with fill
- Engineers told planners it'd sink [6, 8] m
- Planners elected to design for 6 m
- It's sunk 9 m so far and is still sinking

(The operator of the airport denies these media reports)

Variability = aleatory uncertainty

- Arises from natural stochasticity
- Variability arises from
 - spatial variation
 - temporal fluctuations
 - manufacturing or genetic differences
- Not reducible by empirical effort

Incertitude = epistemic uncertainty

- Arises from incomplete knowledge
- Incertitude arises from
 - limited sample size
 - mensurational limits (“measurement error”)
 - use of surrogate data
- Reducible with empirical effort

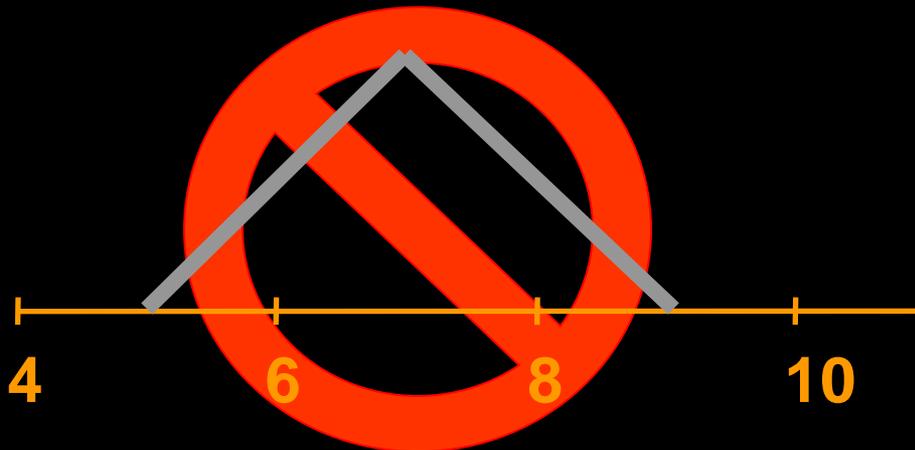
Propagating uncertainty

Suppose

A is in $[2, 4]$

B is in $[3, 5]$

What can be said about the sum $A+B$?



The right answer for engineering is $[5, 9]$

They must be treated *differently*

- Variability should be modeled as randomness with the methods of probability theory
- Incertitude should be modeled as ignorance with the methods of interval analysis
- Imprecise probabilities can do both at once

Incertitude is common in engineering

- Periodic observations

When did the fish in my aquarium die during the night?

- Plus-or-minus measurement uncertainties

Coarse measurements, measurements from digital readouts

- Non-detects and data censoring

Chemical detection limits, studies prematurely terminated

- Privacy requirements

Epidemiological or medical information, census data

- Theoretical constraints

Concentrations, solubilities, probabilities, survival rates

- Bounding studies

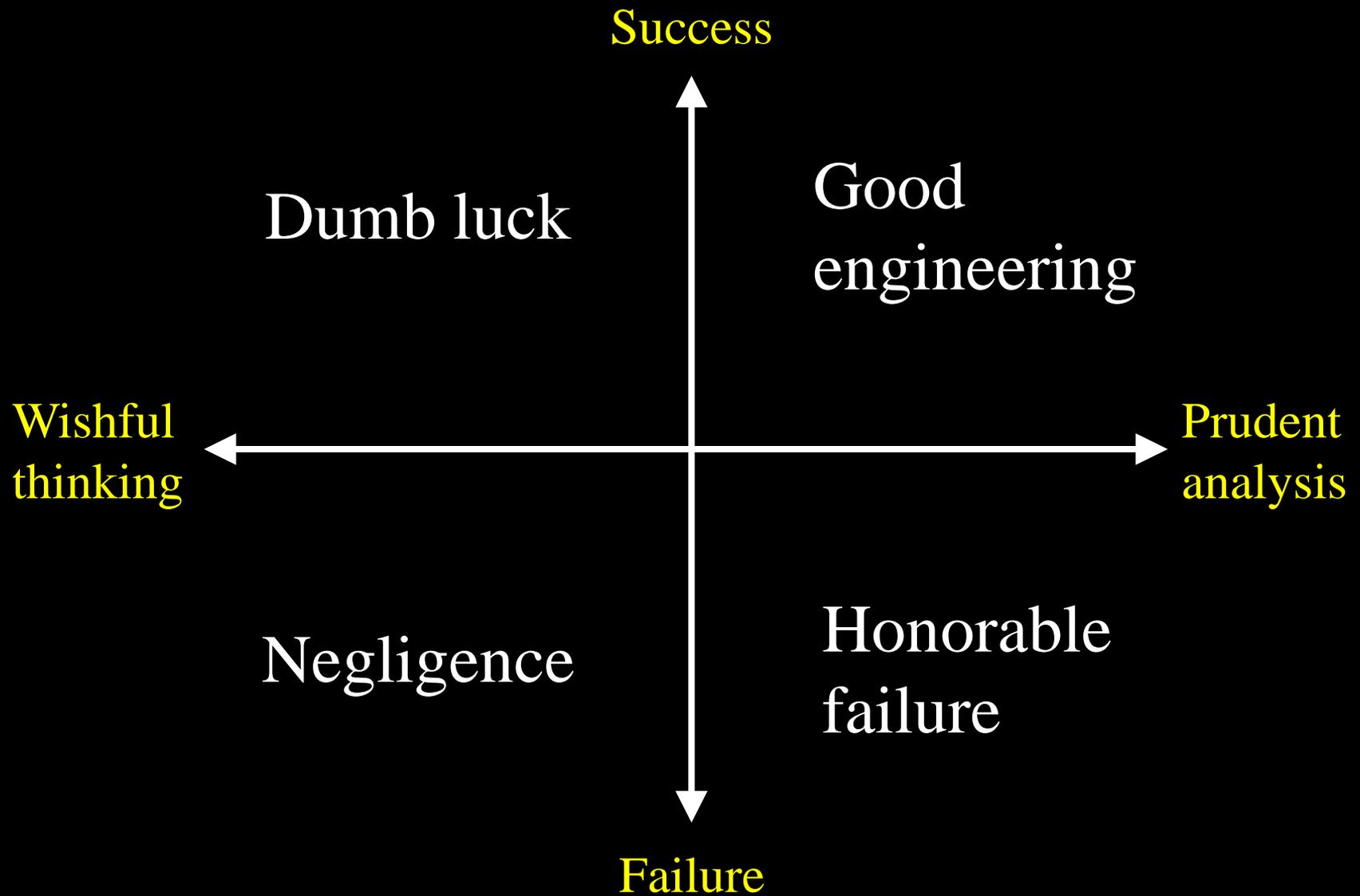
Presumed or hypothetical limits in what-if calculations

Wishful thinking

- Pretending you know the
 - Value
 - Distribution function
 - Dependence
 - Model

when you don't is wishful thinking

- Uncertainty analysis makes a prudent analysis



Traditional uncertainty analyses

- Interval analysis
- Taylor series approximations (delta method)
- Normal theory propagation (ISO/NIST)
- Monte Carlo simulation
- Stochastic PDEs
- Two-dimensional Monte Carlo

Untenable assumptions

- Uncertainties are small
- Distribution shapes are known
- Sources of variation are independent
- Uncertainties cancel each other out
- Linearized models good enough
- Underlying physics is known and modeled

Need ways to relax assumptions

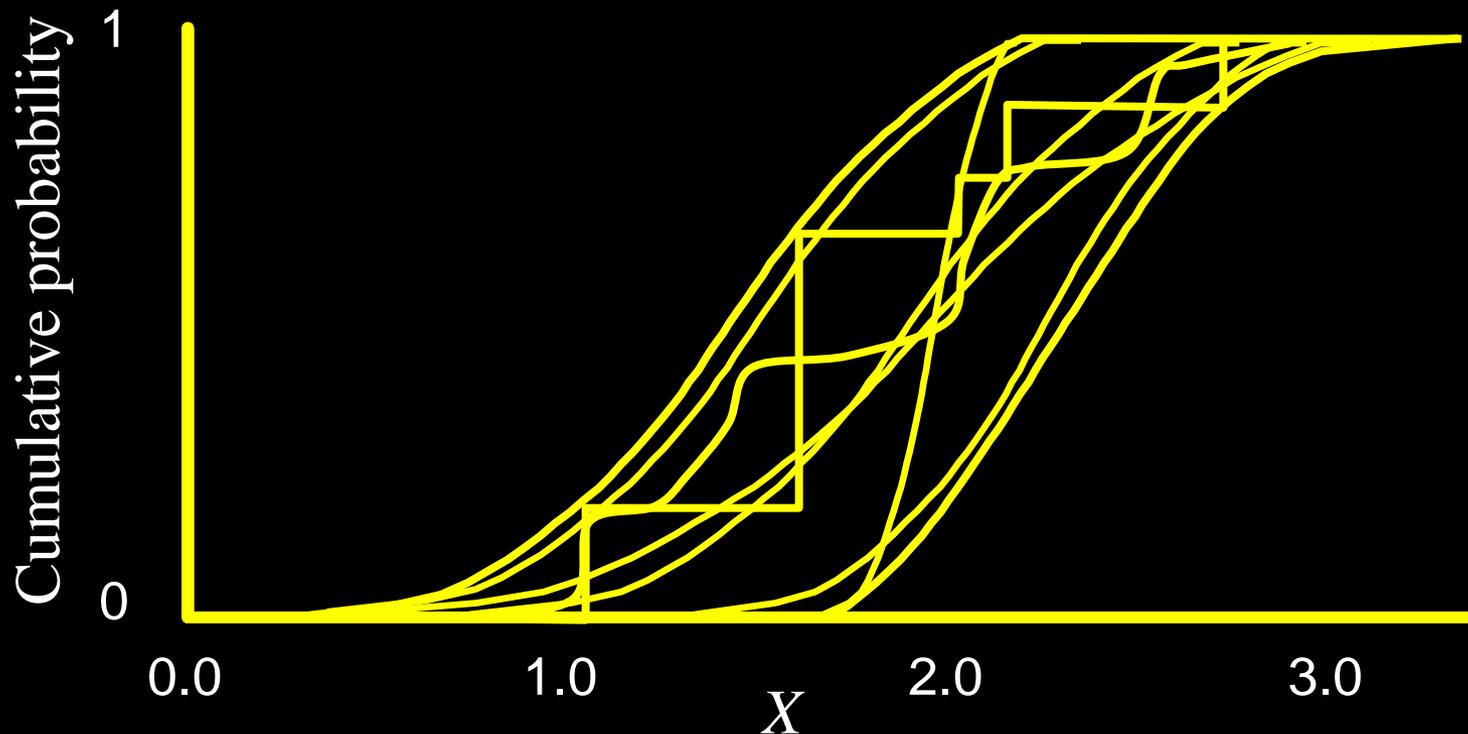
- Hard to say what the distribution is precisely
- Non-independent, or *unknown* dependencies
- Uncertainties that may not cancel
- Possibly large uncertainties
- Model uncertainty

Probability bounds analysis (PBA)

- Sidesteps the major criticisms
 - Doesn't force you to make any assumptions
 - Can use only whatever information is available
- Bridges worst case and probabilistic analysis
- Distinguishes variability and incertitude
- Acceptable to both Bayesians and frequentists

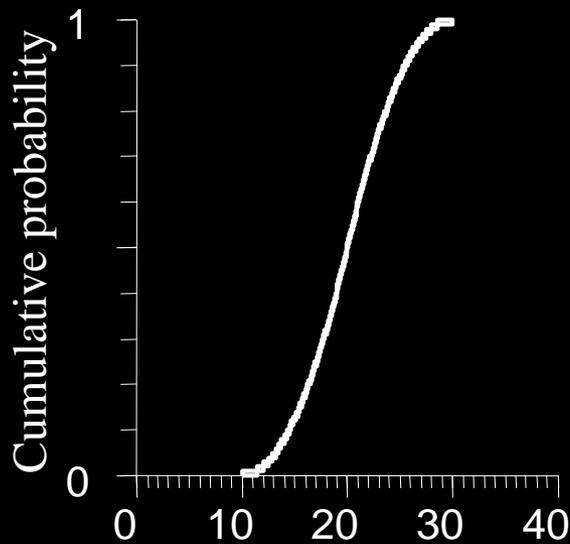
Probability box (p-box)

Interval bounds on a cumulative distribution function

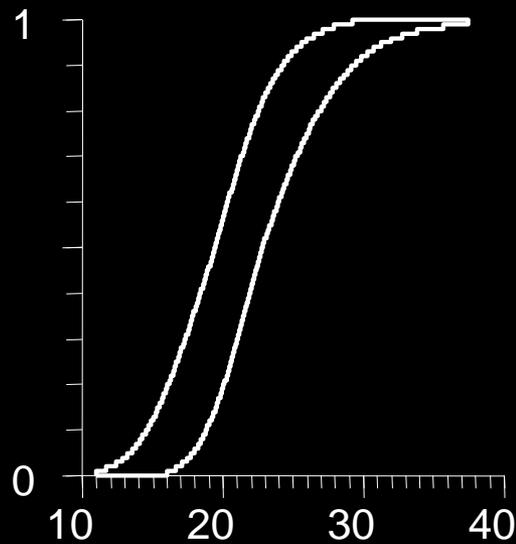


Uncertain numbers

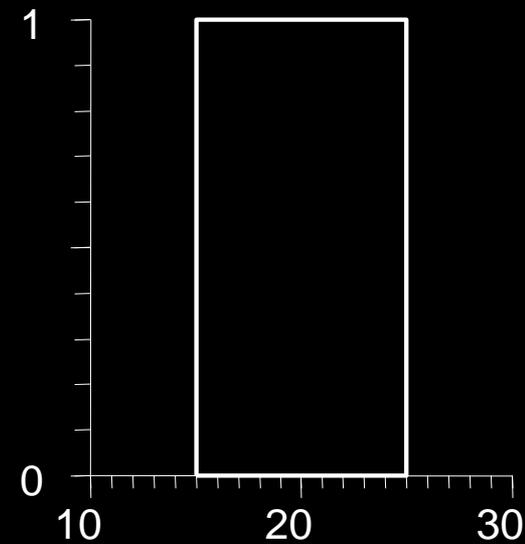
Probability
distribution



Probability
box



Interval



*Not a uniform
distribution*

Uncertainty arithmetic

- We can do math on p-boxes
- When inputs are distributions, the answers conform with probability theory
- When inputs are intervals, the results agree with interval (worst case) analysis

Calculations

- All standard mathematical operations
 - Arithmetic (+, −, ·, /, ^, min, max)
 - Transformations (exp, ln, sin, tan, abs, sqrt, etc.)
 - Magnitude comparisons (<, ≤, >, ≥, ⊆)
 - Other operations (nonlinear ODEs, finite-element methods)
- Faster than Monte Carlo
- Guaranteed to bound the answer
- Optimal solutions often easy to compute

Probability bounds analysis

- Special case of imprecise probabilities
- Addresses many problems in risk analysis
 - Input distributions unknown
 - Imperfectly known correlation and dependency
 - Large measurement error, censoring
 - Small sample sizes
 - Model uncertainty

Better than sensitivity analysis

- Unknown distribution is hard for sensitivity analysis since infinite-dimensional problem
- Analysts usually fall back on a maximum entropy approach, which erases uncertainty rather than propagates it
- Bounding seems very reasonable, so long as it reflects all available information

Example: uncontrolled fire

$$F = A \ \& \ B \ \& \ C \ \& \ D$$

Probability of ignition source

Probability of abundant fuel presence

Probability fire detection not timely

Probability of suppression system failure

Imperfect information

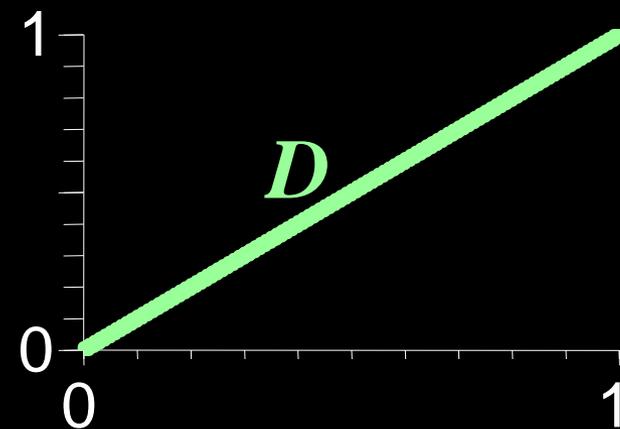
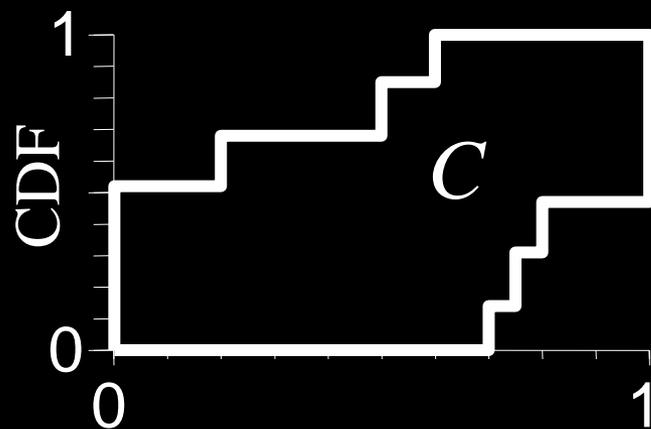
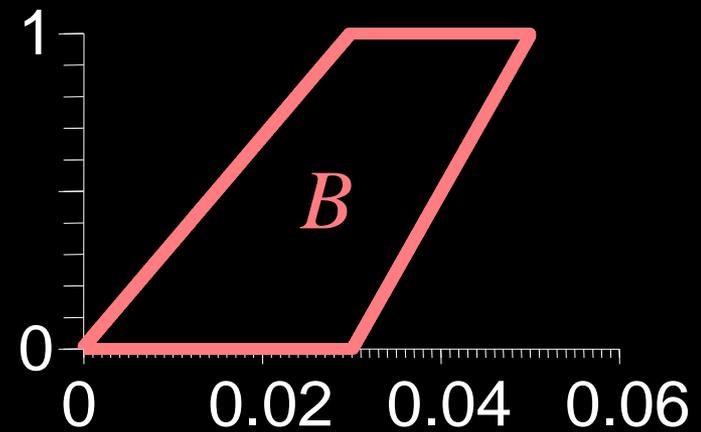
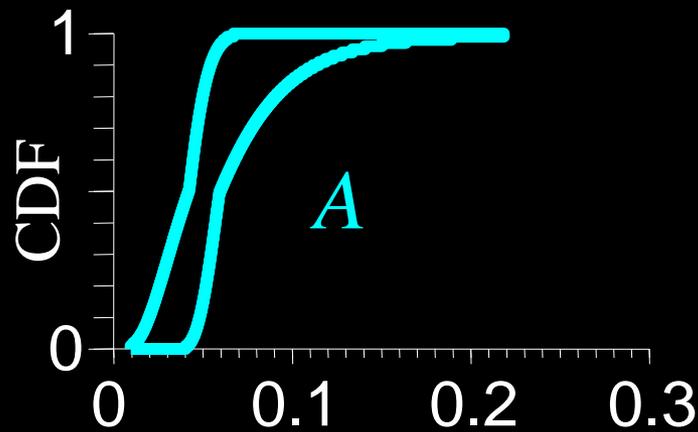
- Calculate A & B & C & D , with partial information:
 - A 's distribution is known, but not its **parameters**
 - B 's parameters known, but not its **shape**
 - C has a small empirical **data set**
 - D is known to be a **precise** distribution
- Bounds assuming independence?
- Without any assumption about dependence?

$A = \{\text{lognormal, mean} = [.05,.06], \text{ variance} = [.0001,.001]\}$

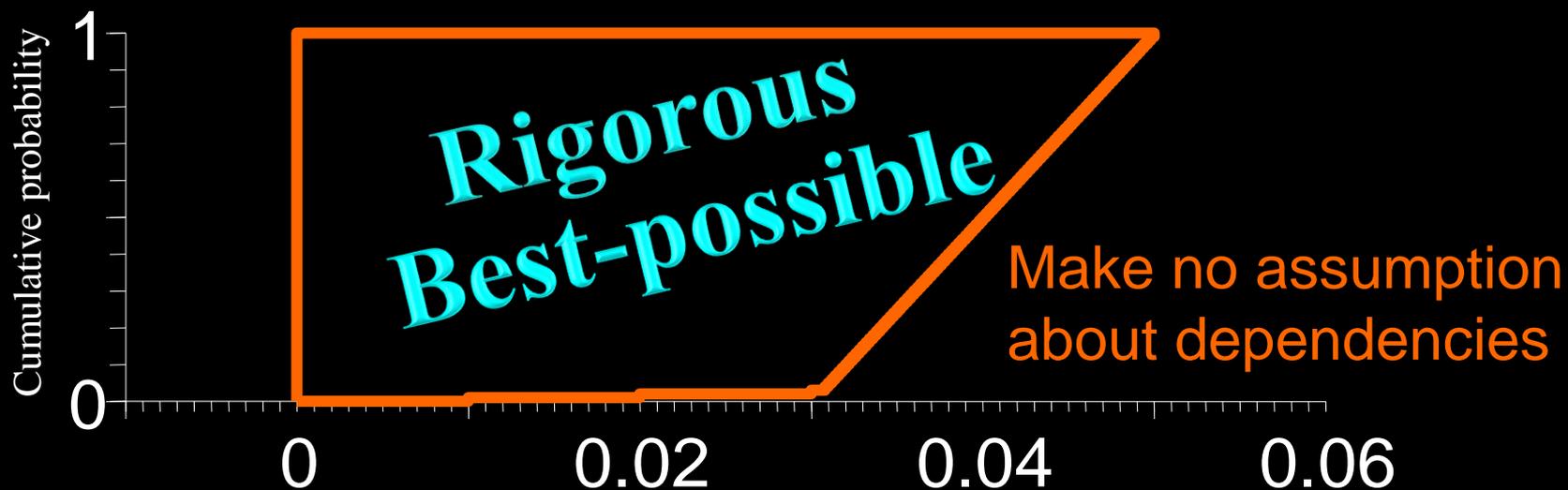
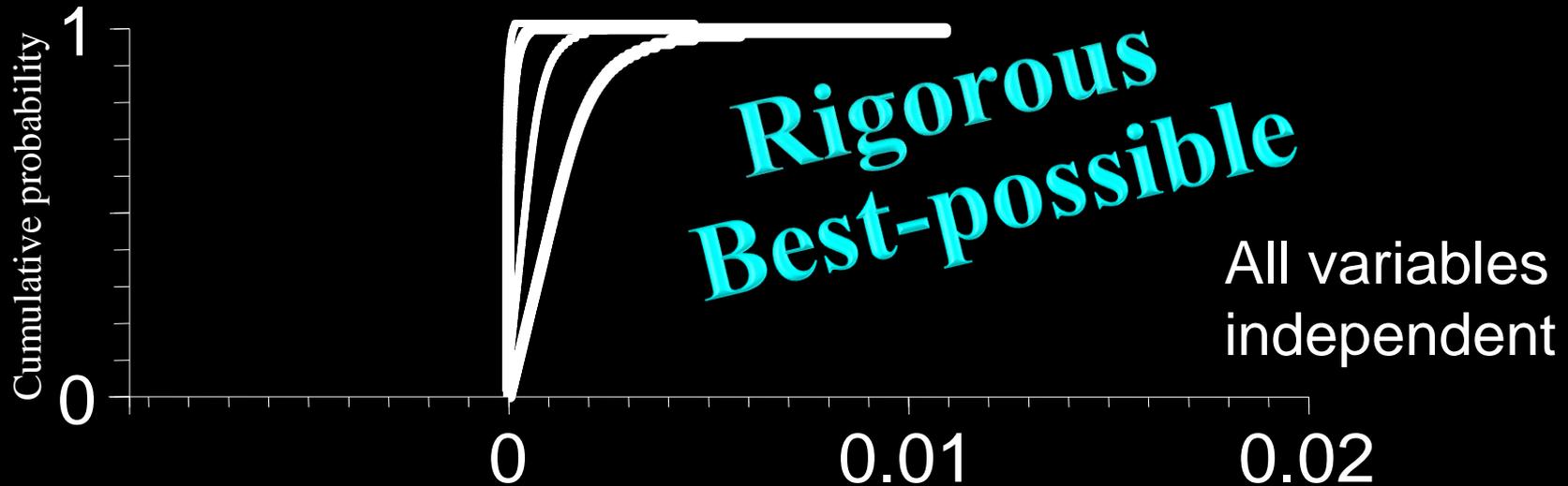
$B = \{\text{min} = 0, \text{max} = 0.05, \text{mode} = 0.03\}$

$C = \{\text{sample data} = 0.2, 0.5, 0.6, 0.7, 0.75, 0.8\}$

$D = \text{uniform}(0, 1)$



Resulting answers



Summary statistics

Independent

| | |
|--------------------|---|
| Range | [0, 0.011] |
| Median | [0, 0.00113] |
| Mean | [0.00006, 0.00119] |
| Variance | [2.9×10^{-9} , 2.1×10^{-6}] |
| Standard deviation | [0.000054, 0.0014] |

No assumptions about dependence

| | |
|--------------------|--------------|
| Range | [0, 0.05] |
| Median | [0, 0.04] |
| Mean | [0, 0.04] |
| Variance | [0, 0.00052] |
| Standard deviation | [0, 0.023] |

How to use the results

When uncertainty makes no difference
(because results are so clear), bounding gives
confidence in the reliability of the decision

When uncertainty swamps the decision

- (i) use other criteria within probability bounds, or
- (ii) use results to identify inputs to study better

Justifying further empirical effort

- If uncertainty is too wide for decisions, and bounds are best possible, more data is needed
- Strong argument for collecting more data

Advantages

- Computationally efficient
 - No simulation or parallel calculations needed
- Fewer assumptions
 - Not just different assumptions, *fewer* of them
 - Distribution-free probabilistic risk analysis
- Rigorous results
 - Built-in quality assurance
 - Automatically verified calculation

Disadvantages

- P-boxes don't say what outcome is most likely
- Hard to get optimal bounds on non-tail risks
- Some technical limits (e.g., sensitive to repeated variables, tricky with black boxes)
- A p-box may not express the tightest possible bounds given all available information (although it often will)

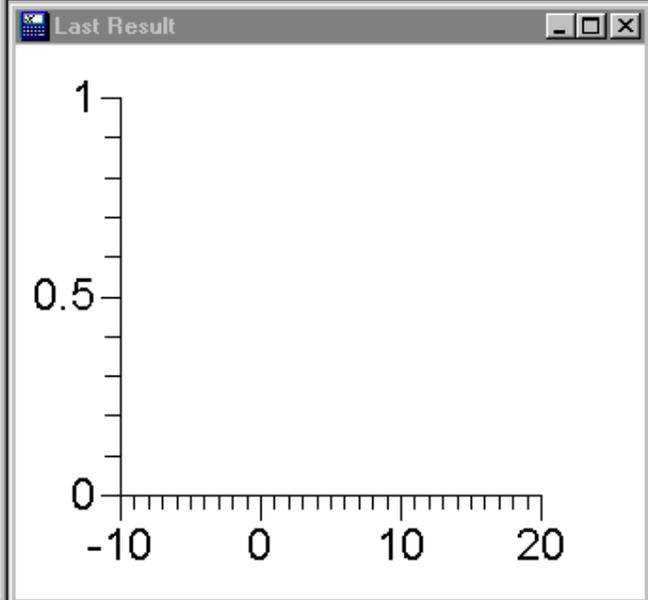
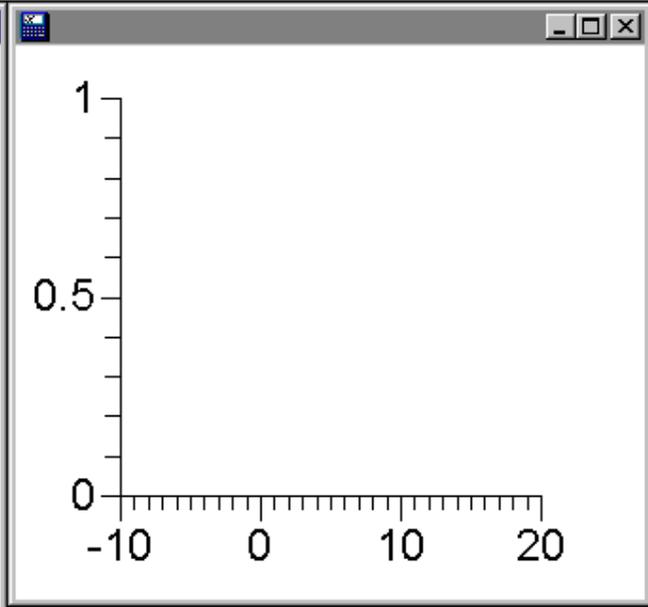
Software

Need β testers

- UC add-in for Excel (NASA, beta 2011)
- RAMAS Risk Calc 4.0 (NIH, commercial)
- Statool (Dan Berleant, freeware)
- Constructor (Sandia and NIH, freeware)
- Pbox.r library for R
- PBDemo (freeware)
- Williamson and Downs (1990)

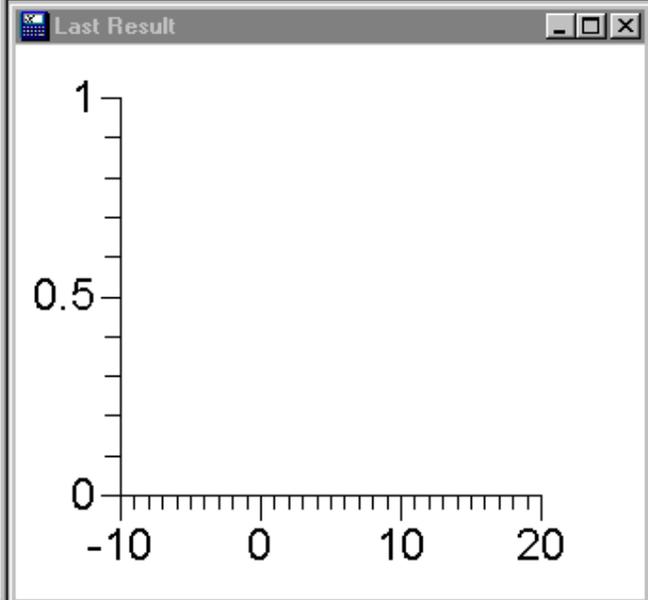
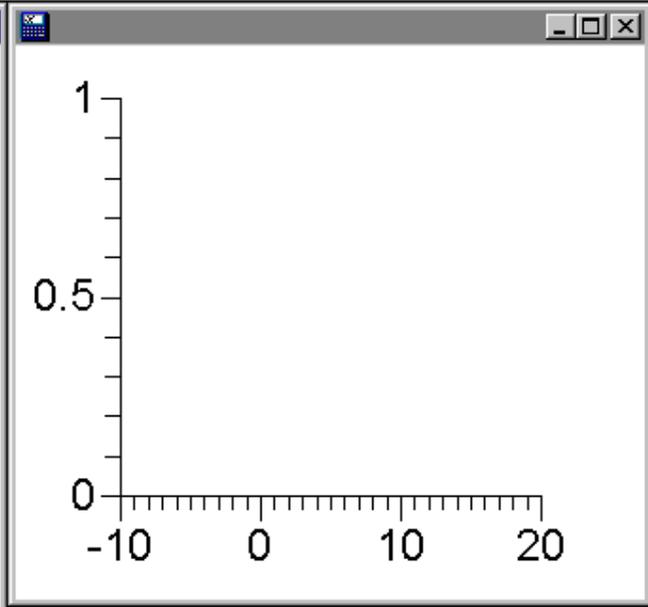


Listener

A large, empty rectangular area with a thin border, intended for displaying the output of a listener.

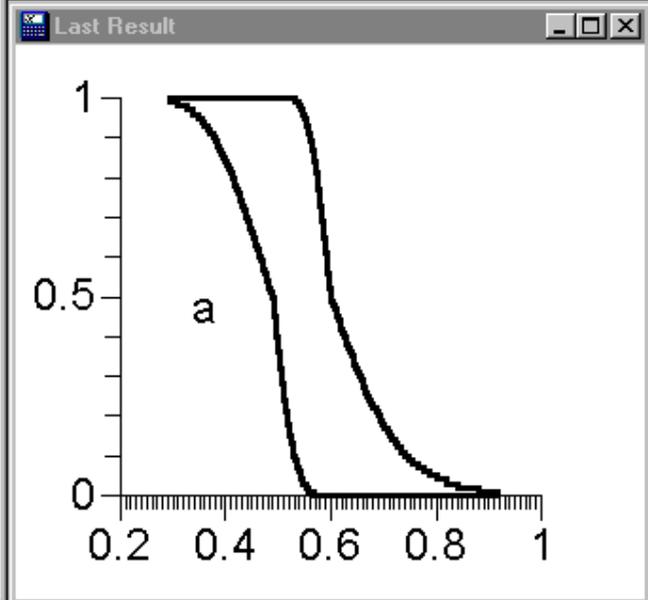
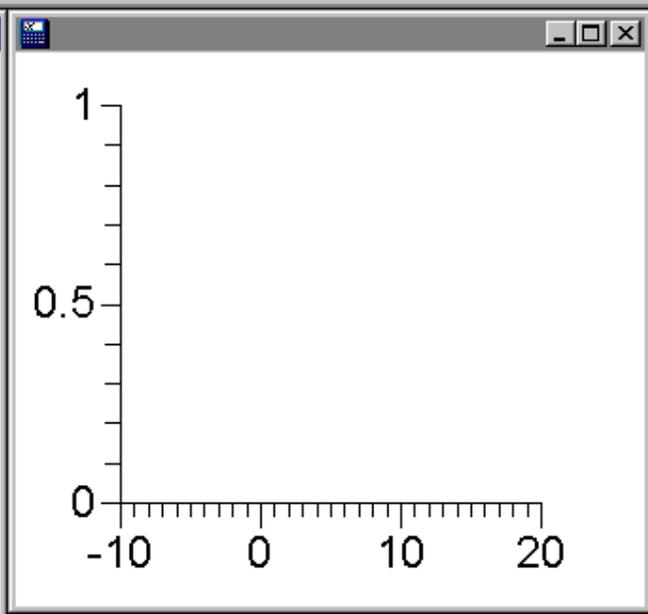


```
Listener  
a = lognormal( [0.5, 0.6], sqrt([0.001, 0.01]) )
```



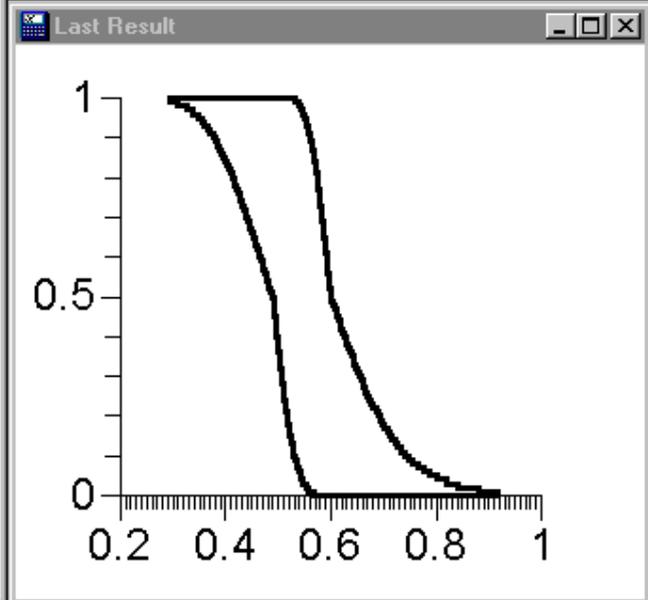
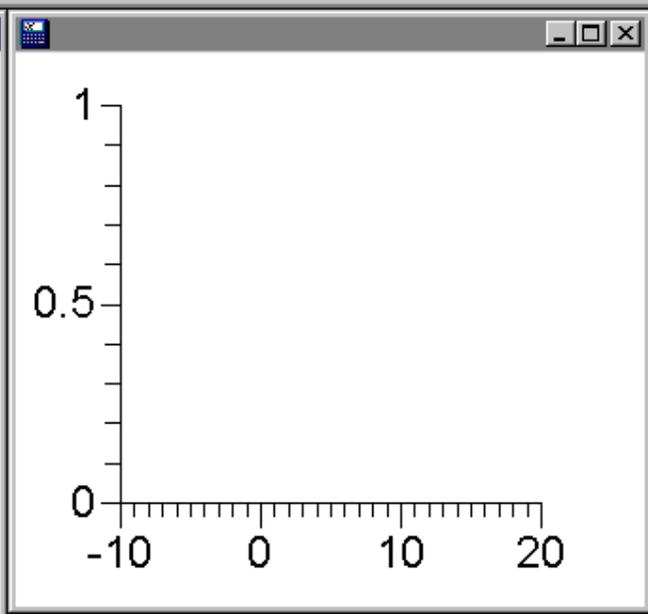


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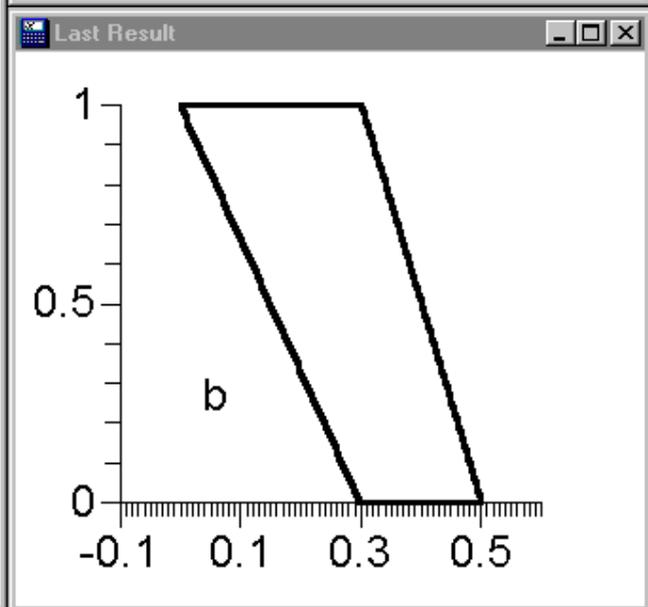
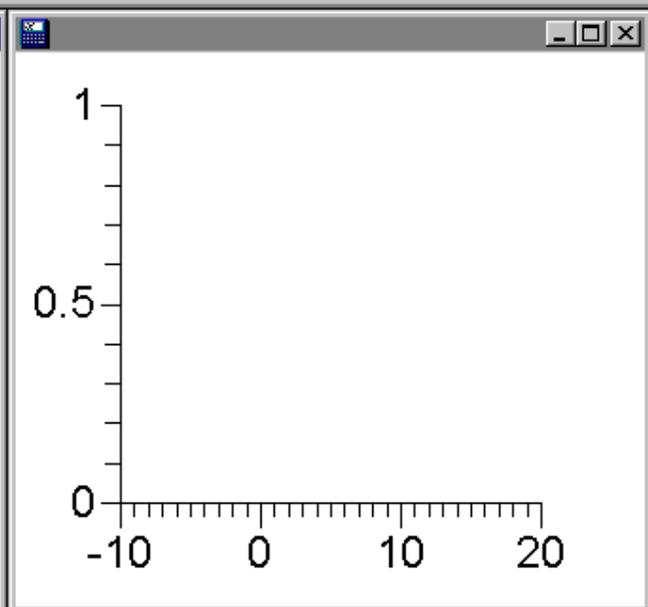


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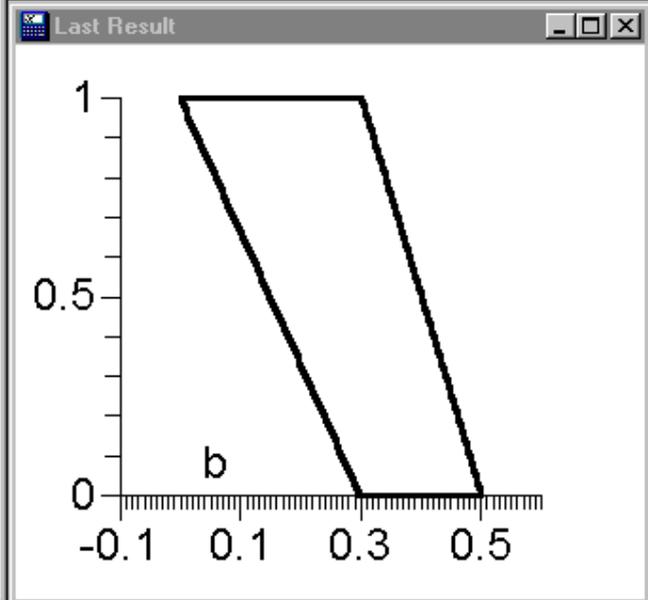
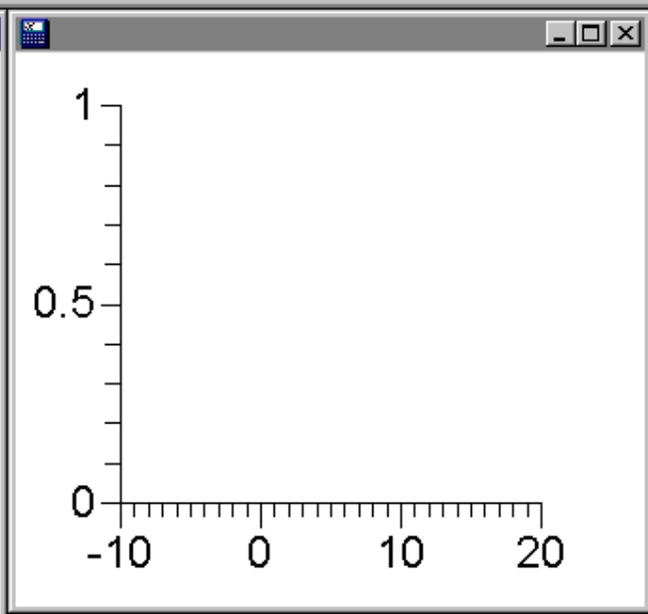


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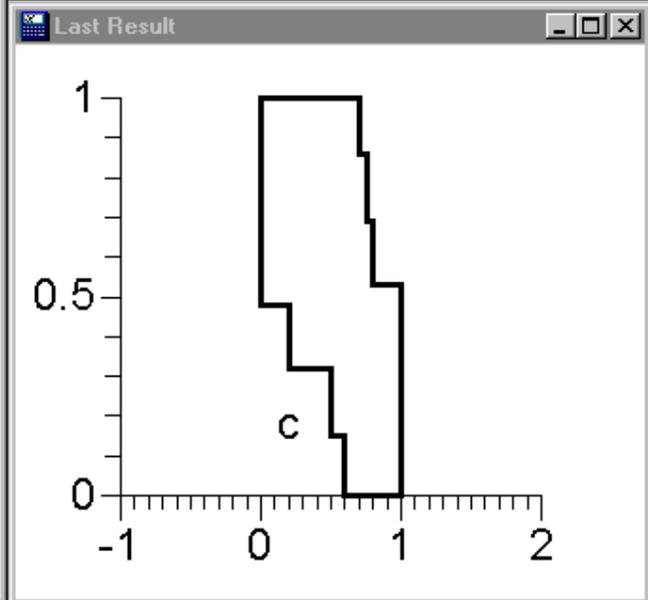
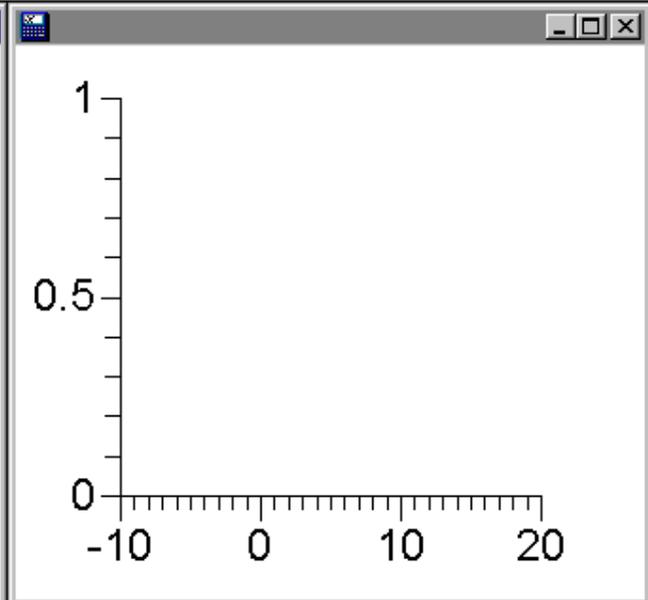


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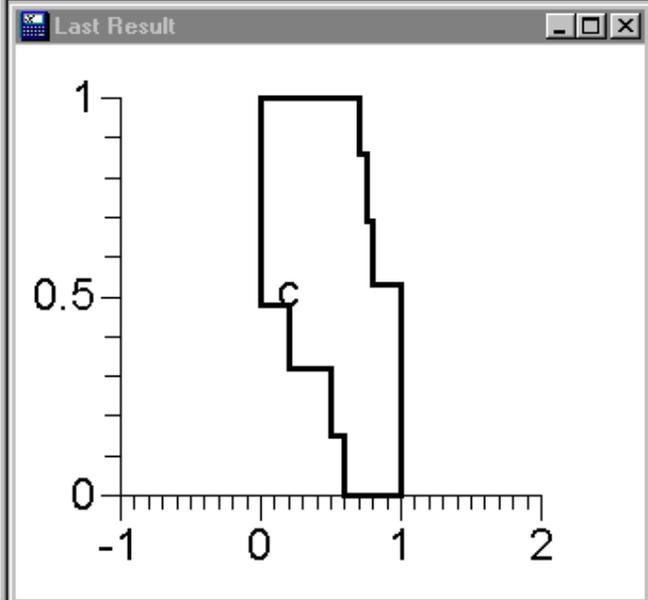
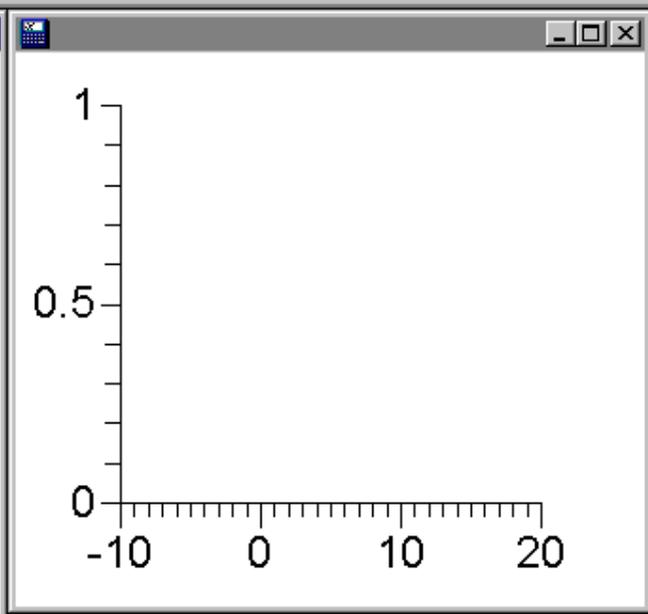


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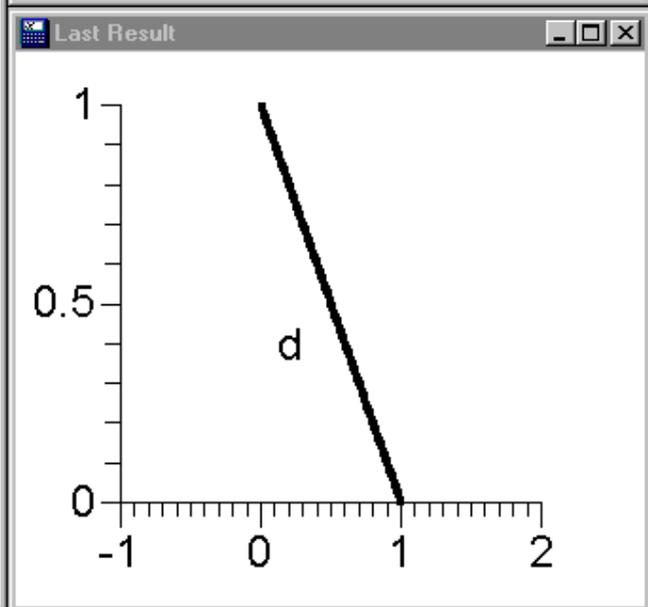
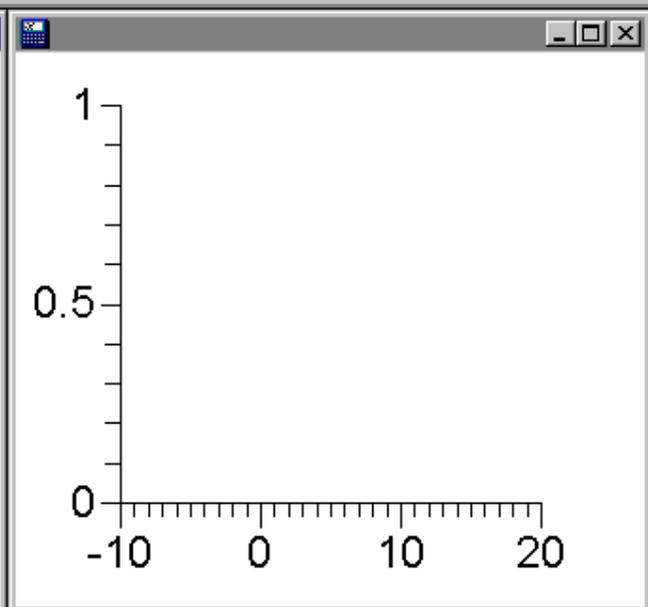


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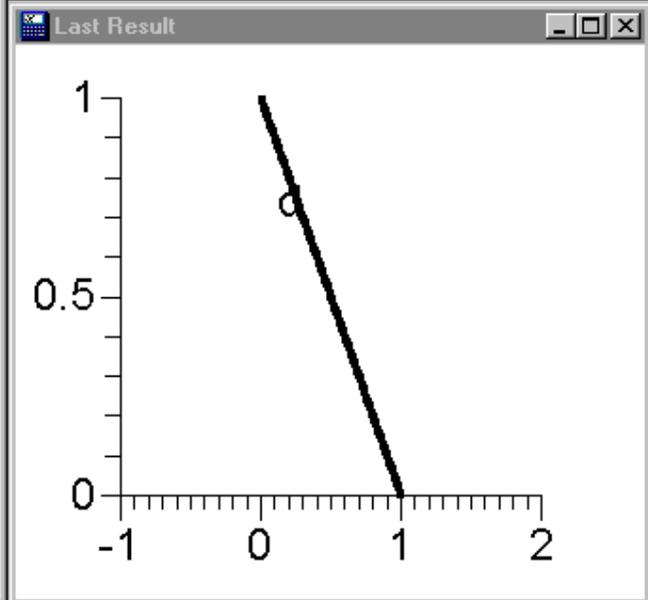
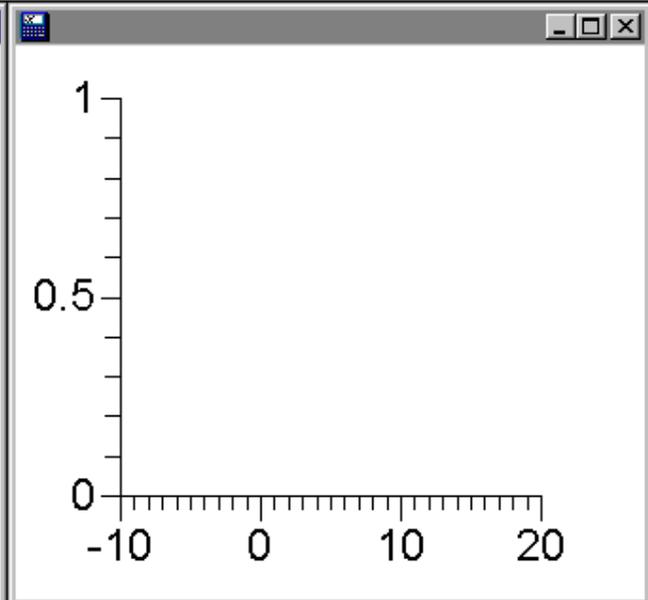


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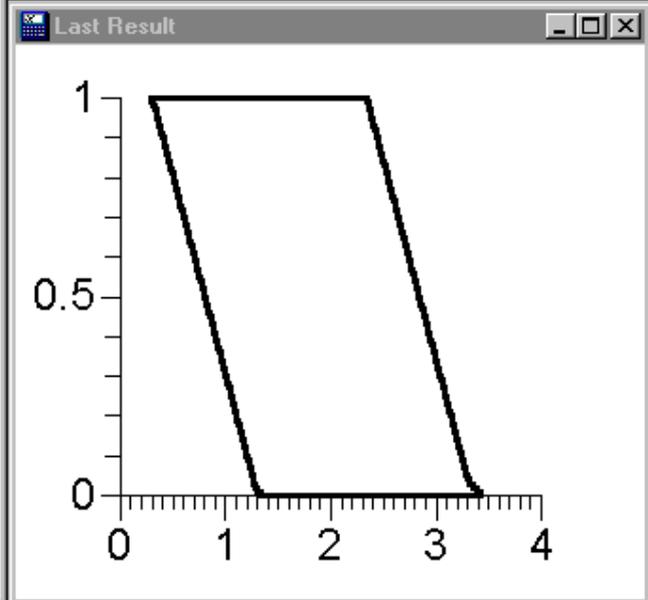
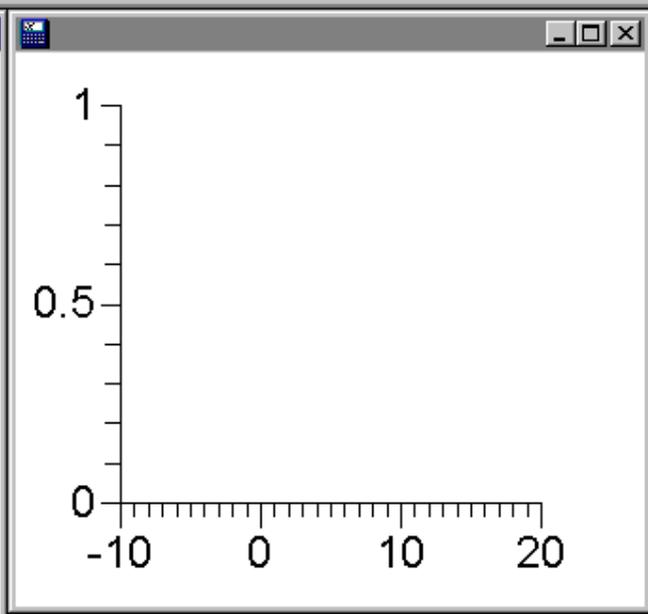


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a + b + c + d|
```





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a + b + c + d  
~(range=[0.294381,3.41774], mean=[1.34,2.4], var=[0,1.07]  
|
```





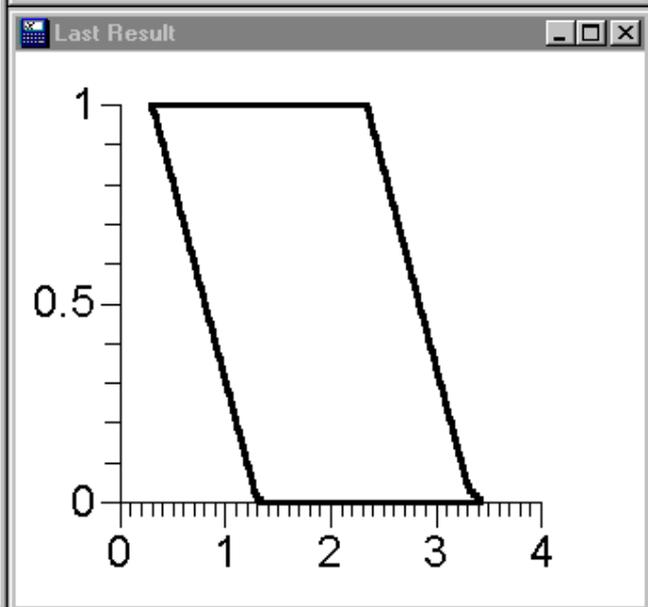
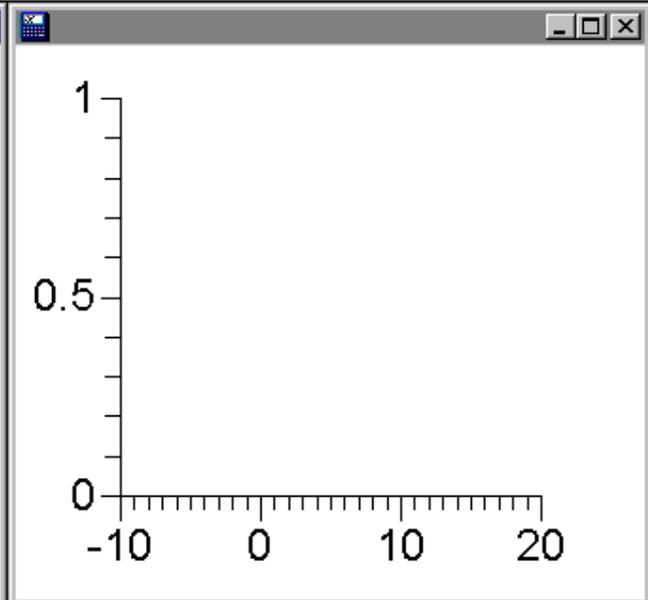
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a + b + c + d
  ~(range=[0.294381,3.41774], mean=[1.34,2.4], var=[0,1.07]

a |+| b |+| c |+| d|

```





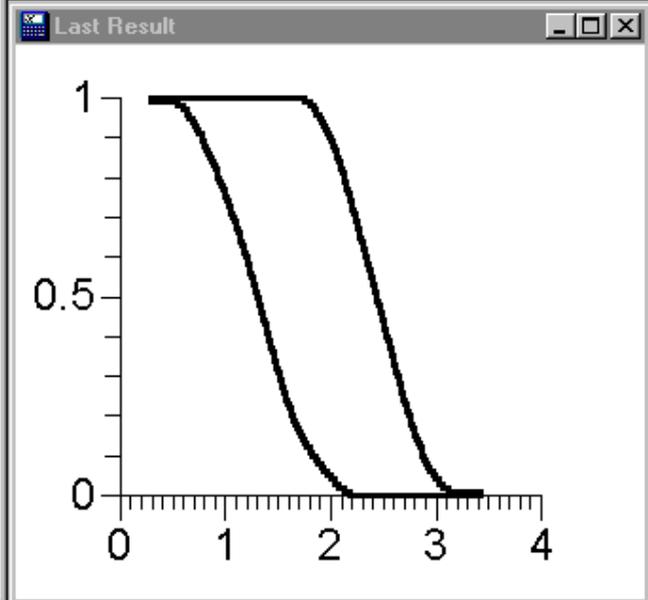
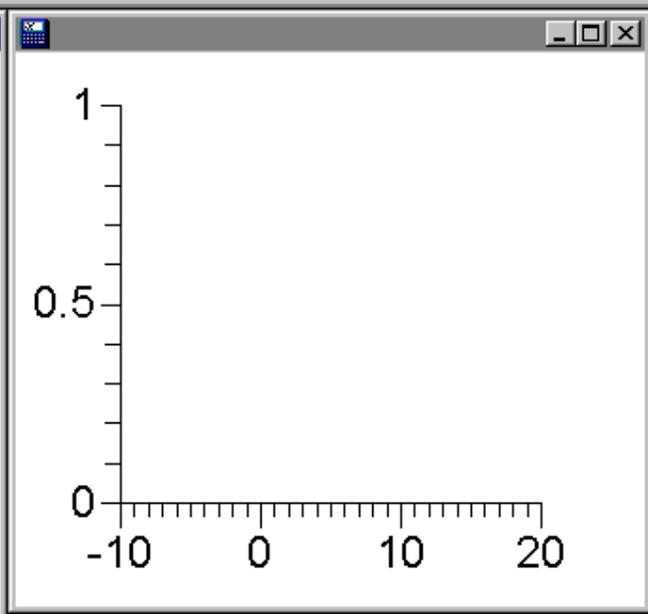
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a |+| b |+| c |+| d
  ~(range=[0.294381,3.41774], mean=[1.34,2.4], var=[0.084
|

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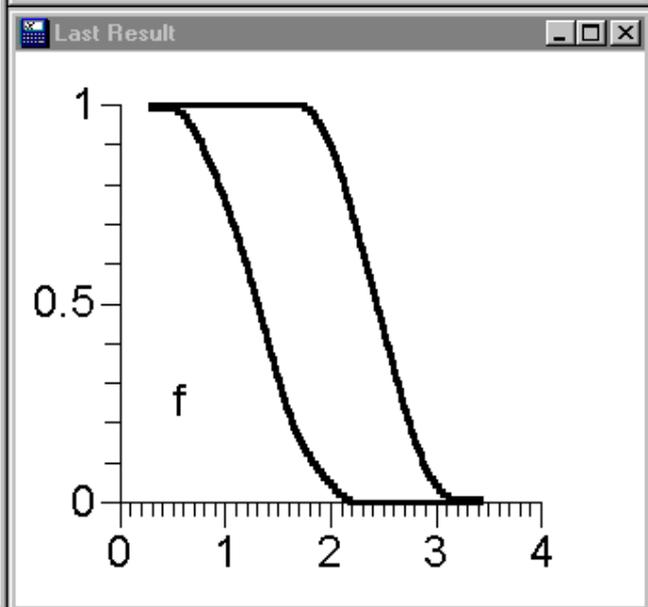
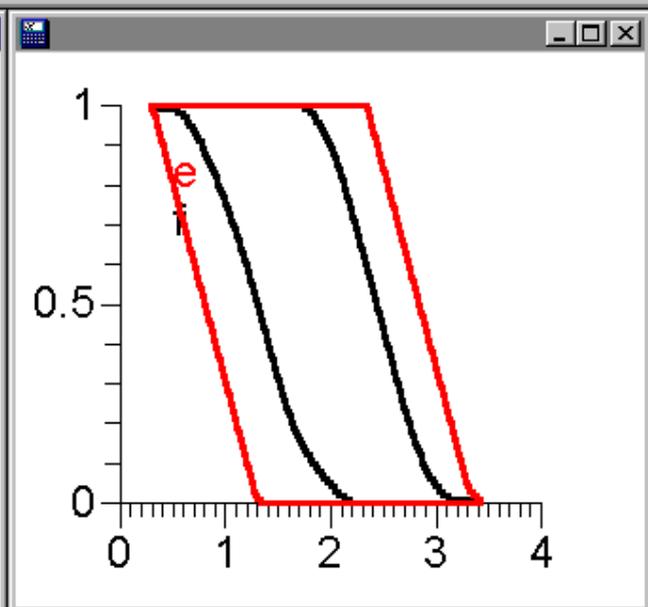
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d = uniform(0, 1)

e = a + b + c + d
  ~(range=[0.294381,3.41774], mean=[1.34,2.4], var=[0,1.0])

f = a |+| b |+| c |+| d
  ~(range=[0.294381,3.41774], mean=[1.34,2.4], var=[0.08])

show e in red
show f
|

```



Diverse applications

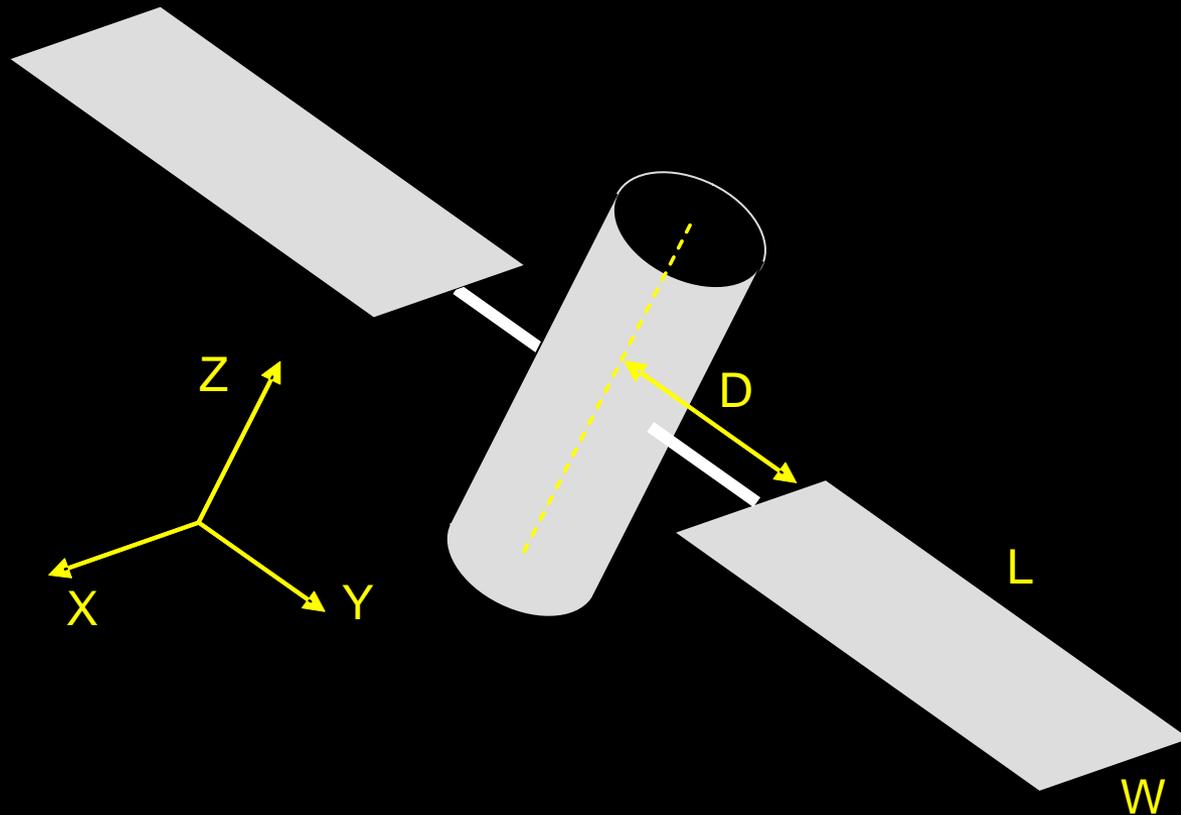
- Superfund risk analyses
- Conservation biology extinction/reintroduction
- Occupational exposure assessment
- Food safety
- Chemostat dynamics
- Global climate change forecasts
- Safety of engineered systems
- Engineering design

Case study:

Spacecraft design under
mission uncertainty

Mission

Deploy satellite carrying a large optical sensor



Microsoft Excel - SMAD Design WS-340km-detailed-5.xls

File Edit View Insert Format Tools Data Window Help Acrobat

Times New Roman 16 B I U \$ %

A1 = Design Sheet Navigator

| | A | B | C | D | E | F | G | H | I | J | K | L | M | N | O | P | Q | R | S | |
|----|--|---|-----------------|---|-------------------------------------|-------------------------------------|--------------------------------------|---|---|-------------------------------|------------------------------|----------------------|---|--------------|---|---|---|---|---|--|
| 1 | Design Sheet Navigator | | | | | | | | | | | | | | | | | | | |
| 2 | | | | | | | | | | | | | | | | | | | | |
| 3 | Orbit Analysis | | | | | Observation Payload Analysis | | | | | Spacecraft Subsystems | | | | | | | | | |
| 4 | Orbit dynamics | | Dynamics | | Subject and EM Spectrum | | Spectrum | | | Preliminary Sizing | | Prelim Sizing | | | | | | | | |
| 5 | Mission geometry | | Geometry | | Optics | | Optics | | | | | | | | | | | | | |
| 6 | Orbit maneuvers and maintenance | | Maneuvers | | Sizing | | Sizing | | | Attitude Control | | | | | | | | | | |
| 7 | Delta-V & geometry budgets | | Budgets | | | | | | | Torques | | Att - Torques | | | | | | | | |
| 8 | | | | | | | | | | Sizing | | Att - Sizing | | | | | | | | |
| 9 | | | | | Launch Vehicle Information | | Launch Vehicle | | | Communications | | | | | | | | | | |
| 10 | | | | | | | | | | Uplink | | Comm - Uplink | | | | | | | | |
| 11 | | | | | Transfer Vehicle Information | | Transfer Vehicle | | | Downlink | | Comm - Downlink | | | | | | | | |
| 12 | Cost Estimation | | Mission Inputs | | Cost Inputs | | | | | Power | | | | | | | | | | |
| 13 | | | | | | | Mission Operations Complexity | | | Ops Complexity | | Solar array analysis | | Solar Arrays | | | | | | |
| 14 | Space Segment | | | | | | | | | Secondary battery analysis | | Batteries | | | | | | | | |
| 15 | USCM 7th Edition (SMAD Table 20-4 & 20-5, p 795-796) | | | | | | | | | Other primary power sources | | Other Sources | | | | | | | | |
| 16 | | | USCM | | | | | | | Propulsion | | | | | | | | | | |
| 17 | | | | | | | | | | Sizing | | Prop - Sizing | | | | | | | | |
| 18 | SSCM (SMAD Table 20-6, p 797) | | SSCM | | | | | | | Thermodynamics | | Prop - Thermo | | | | | | | | |
| 19 | | | | | | | | | | Storage and Feed | | Prop - Storage | | | | | | | | |
| 20 | SSCM 7.4 (RSMC Table 8-4, p 271) | | | | | | | | | Structures | | | | | | | | | | |
| 21 | Spacecraft Bus Cost | | SSCM 7.4 Bus | | | | | | | Monocoque | | Structure - Mono | | | | | | | | |
| 22 | System Cost | | SSCM 7.4 Sys | | | | | | | Semi-monocoque | | Structure - Semi | | | | | | | | |
| 23 | | | | | | | | | | Thermal Control | | | | | | | | | | |
| 24 | SSCM 8.0 (RSMC Table 8-5, p 272) | | | | | | | | | Spherical spacecraft analysis | | Thermal - Sphere | | | | | | | | |
| 25 | Spacecraft Bus Cost | | SSCM 8.0 Bus | | | | | | | Solar array analysis | | Thermal - Solar | | | | | | | | |
| 26 | System Cost | | SSCM 8.0 Sys | | | | | | | System Sizing Summary | | Sizing Summary | | | | | | | | |
| 27 | | | | | | | | | | | | | | | | | | | | |
| 28 | Cost Model Comparison | | Cost Comparison | | | | | | | | | | | | | | | | | |
| 29 | | | | | | | | | | | | | | | | | | | | |
| 30 | Lifecycle Cost | | Lifecycle Cost | | | | | | | | | | | | | | | | | |
| 31 | | | | | | | | | | | | | | | | | | | | |

Design Sheet Navigator / Constants / Orbits - Dynamics / Orbits - Geometry / Orbits - Maneuvers / Orbits - Budgets / Subject - EM Spectr

Wertz and Larson (1999) *Space Mission Analysis and Design (SMAD)*. Kluwer.

| | A | B | C | D | E | F | G | H | I | J | K | L |
|----|---|--|-----------|-------------------|------------------|-----------|------------------------------|---|---|---|---|---|
| 1 | Return to Navigator | Attitude Control - Torque Estimates | | | | | | | | | | |
| 2 | <i>(All information on this sheet is contained in the block from Cell A1 to Cell Q27)</i> | | | | | | | | | | | |
| 3 | | | | | | | | | | | | |
| 4 | Orbit characteristics | | | | | | Environmental torques | | | | | |
| 5 | Altitude | | 340.000 | km | Gravity gradient | 1.794E-03 | N-m | | | | | |
| 6 | Satellite velocity | | 7.703 | km/s | Solar radiation | 2.565E-05 | N-m | | | | | |
| 7 | | | | | Magnetic | 5.250E-05 | N-m | | | | | |
| 8 | Atmospheric density | | 1.983E-11 | kg/m ³ | Aerodynamic | 6.473E-03 | N-m | | | | | |
| 9 | Sun incidence angle | | 0.00 | deg | Total (RSS) | 6.717E-03 | N-m | | | | | |
| 10 | Maximum deviation from local vertical | | 10.00 | deg | | | | | | | | |
| 11 | | | | | | | | | | | | |
| 12 | | | | | | | | | | | | |
| 13 | Spacecraft characteristics | | | | | | Slewing torque | | | | | |
| 14 | Largest moment of inertia | 7315.000 | 7315.000 | kg-m ² | | | | | | | | |
| 15 | Smallest moment of inertia | 4655.000 | 4655.000 | kg-m ² | | | | | | | | |
| 16 | Projected surface area | 14.063 | 14.063 | m ² | | | | | | | | |
| 17 | Moment arm for solar radiation torques | | 0.250 | m | | | | | | | | |
| 18 | Moment arm for aerodynamic torques | | 0.250 | m | | | | | | | | |
| 19 | | | | | | | | | | | | |
| 20 | Drag coefficient | | 3.13 | | | | | | | | | |
| 21 | Surface reflectivity | | 0.60 | | | | | | | | | |
| 22 | Residual dipole | | 1.00 | A-m ² | | | | | | | | |
| 23 | | | | | | | | | | | | |
| 24 | | | | | | | | | | | | |
| 25 | Slew characteristics | | | | | | | | | | | |
| 26 | Maximum slewing angle | 38.00 | 38.00 | deg | | | | | | | | |
| 27 | Minimum maneuver time | 760.00 | 760.00 | sec | | | | | | | | |
| 28 | | | | | | | | | | | | |
| 29 | | | | | | | | | | | | |

Typical subsystems

Attitude control

Command data systems

Configuration

Cost

Ground systems

Instruments

Mission design

Power

Program management

Propulsion

Science

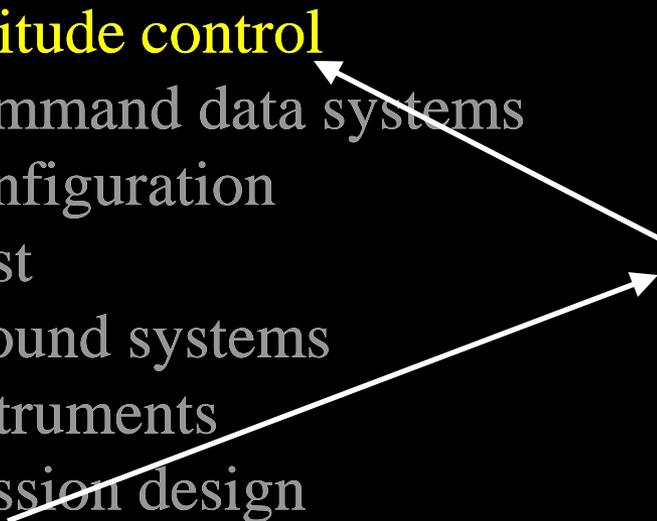
Solar array

Systems engineering

Telecommunications – System

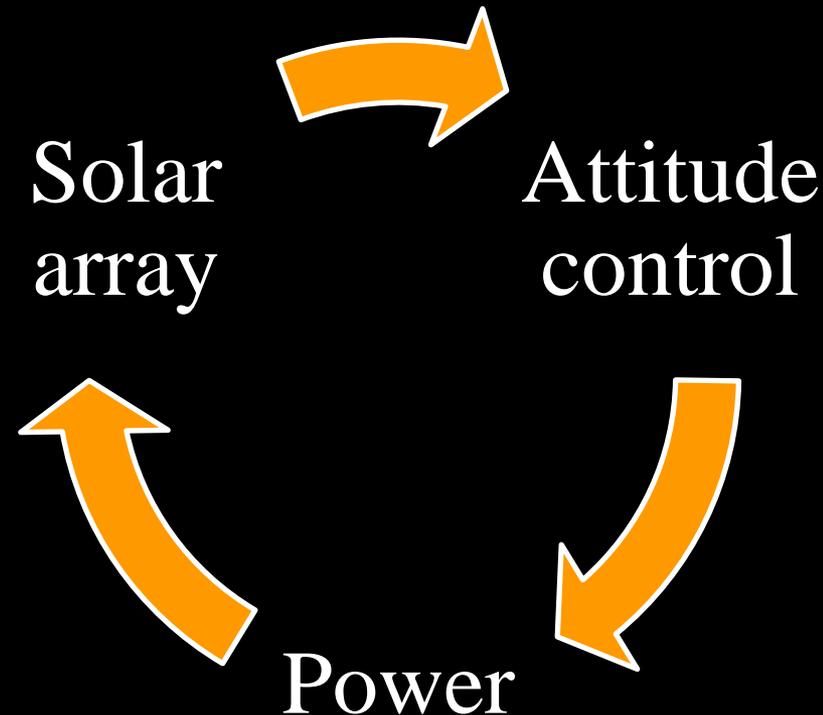
Telecommunications – Hardware

Thermal control



Demonstrations

- Calculations within a single subsystem (ACS)
- Calculations within linked subsystems



Attitude control subsystem (ACS)

- 3 reaction wheels
- Design problem: solve for h
 - Required angular momentum
 - Needed to choose reaction wheels
- Mission constraints
 - $\Delta t_{\text{orbit}} = 1/4$ orbit time
 - $\theta_{\text{slew}} = \text{max slew angle}$
 - $\Delta t_{\text{slew}} = \text{min maneuver time}$
- Inputs from other subsystems
 - $I, I_{\text{max}}, I_{\text{min}} = \text{inertial moment}$
 - Depend on solar panel size, which depends on power needed, so on h

$$h = \tau_{\text{tot}} \times \Delta t_{\text{orbit}}$$

$$\tau_{\text{tot}} = \tau_{\text{slew}} + \tau_{\text{dist}}$$

$$\tau_{\text{slew}} = \frac{4\theta_{\text{slew}}}{\Delta t_{\text{slew}}} I$$

$$\tau_{\text{dist}} = \tau_g + \tau_{\text{sp}} + \tau_m + \tau_a$$

$$\tau_g = \frac{3\mu}{2(R_E + H)^3} |I_{\text{max}} - I_{\text{min}}| \sin(2\theta)$$

$$\tau_{\text{sp}} = L_{\text{sp}} \frac{F_s}{c} A_s (1 + q) \cos(i)$$

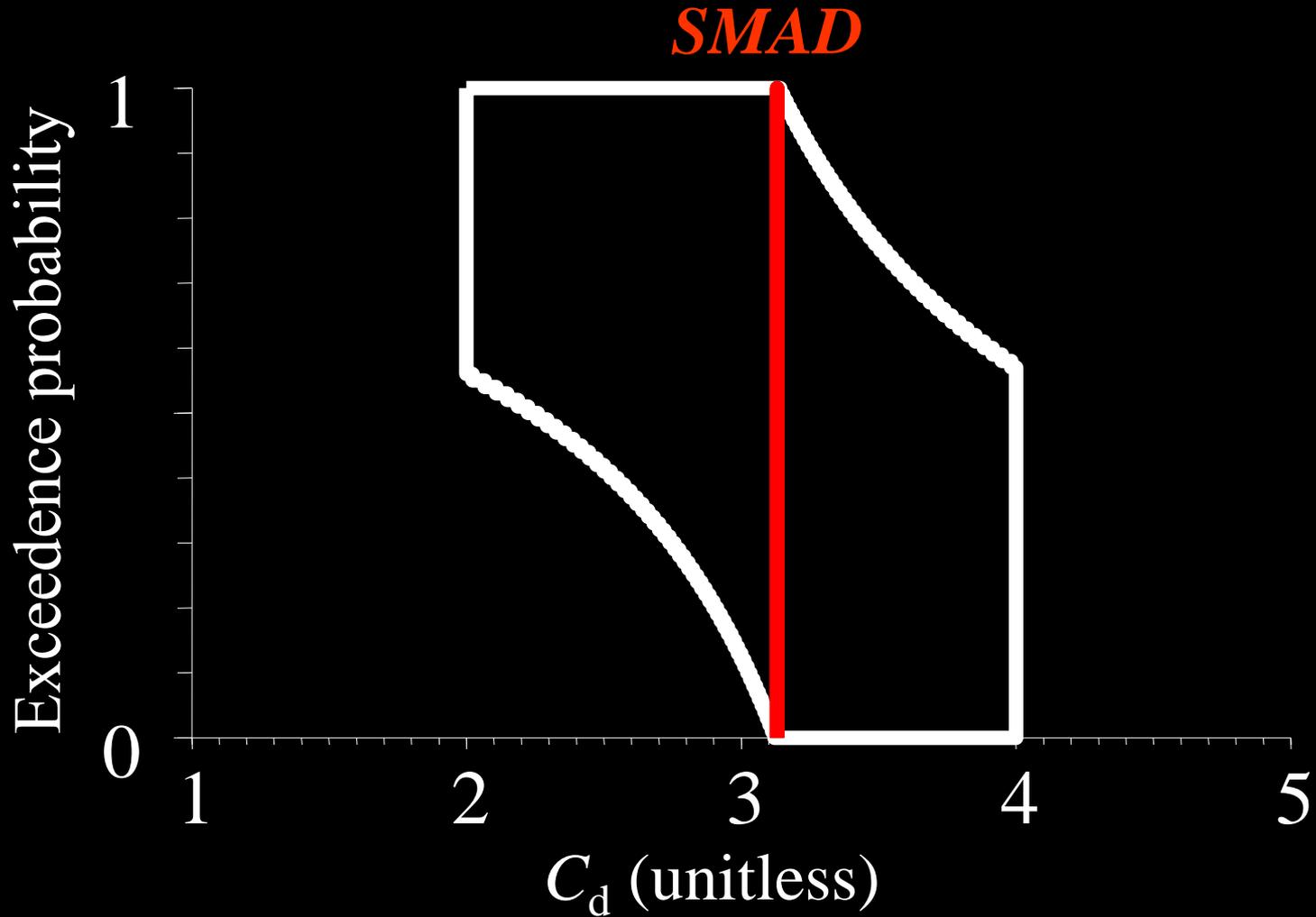
$$\tau_m = \frac{2MD}{(R_E + H)^3}$$

$$\tau_a = \frac{1}{2} L_a \rho C_d A V^2$$

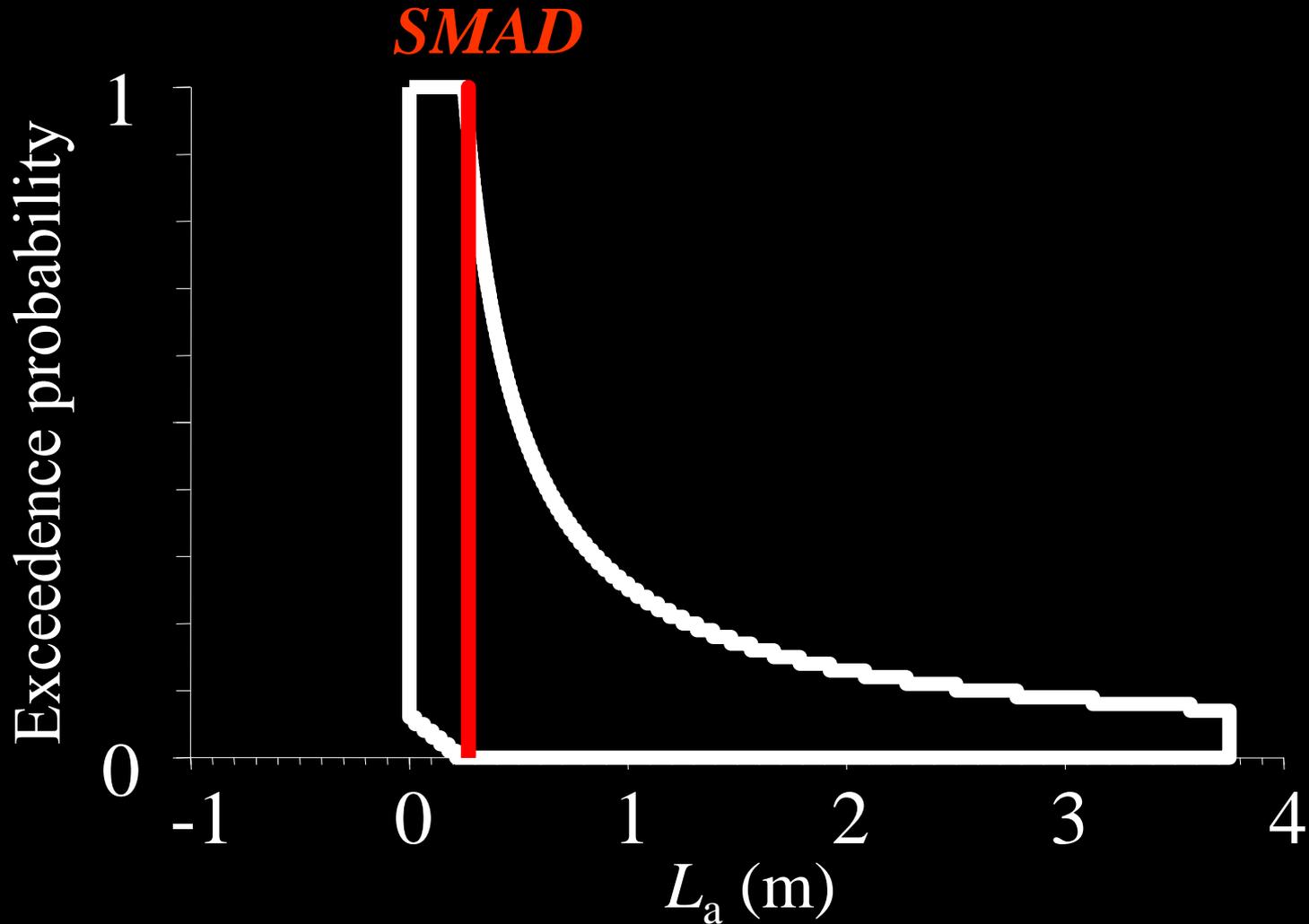
Attitude control input variables

| Symbol | Unit | Variable | Type | Value | SMAD |
|--------------------|--------------------------------|--|----------|-------------------------------|-------------------|
| C_d | unitless | Drag coefficient | p-box | range=[2,4] mean=3.13 | 3.13 |
| L_a | m | Aerodynamic drag torque moment | p-box | range=[0,3.75] mean=0.25 | 0.25 |
| L_{sp} | m | Solar radiation torque moment | p-box | range=[0,3.75] mean=[0.25] | 0.25 |
| D_r | A m ² | Residual dipole | interval | [0,1] | 1 |
| i | degrees | Sun incidence angle | interval | [0,90] | 0 |
| ρ | kg m ³ | Atmospheric density | interval | [3.96e-12, 9.9e-11] | 1.98e-11 |
| θ | degrees | Major moment axis deviation from nadir | interval | [10,19] | 10 |
| q | unitless | Surface reflectivity | interval | [0.1,0.99] | 0.6 |
| I_{min} | kg m ² | Minimum moment of inertia | interval | [4655] | 4655 |
| I_{max} | kg m ² | Maximum moment of inertia | interval | [7315] | 7315 |
| μ | m ³ s ⁻² | Earth gravity constant | point | 3.98e14 | 3.98e14 |
| A | m ² | Area in the direction of flight | point | 3.75 ² | 3.75 ² |
| RE | km | Earth radius | point | 6378.14 | 6378.14 |
| H | km | Orbit altitude | point | 340 | 340 |
| F_s | W m ⁻² | Average solar flux | point | 1367 | 1367 |
| θ_{slew} | degrees | Maximum slewing angle | point | 38 | 38 |
| c | m s ⁻¹ | Light speed | point | 2.9979e8 | 2.9979e8 |
| M | A m ² | Earth magnetic moment | point | 7.96e22 | 7.96e22 |
| Δt_{slew} | s | Minimum maneuver time | point | 760 | 760 |
| A_s | m ² | Area reflecting solar radiation | point | 3.75 ² | 3.75 ² |
| Δt_{orbit} | s | Quarter orbit period | point | 1370 | 1370 |

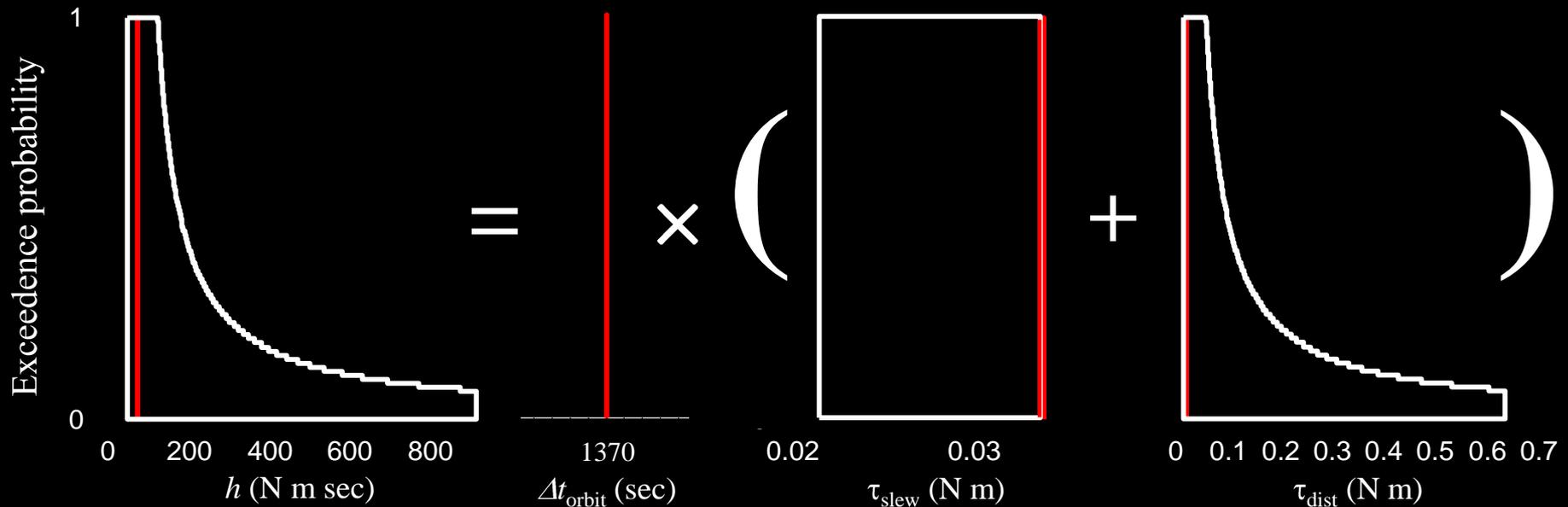
Coefficient of drag, C_d



Aerodynamic drag torque moment, L_a

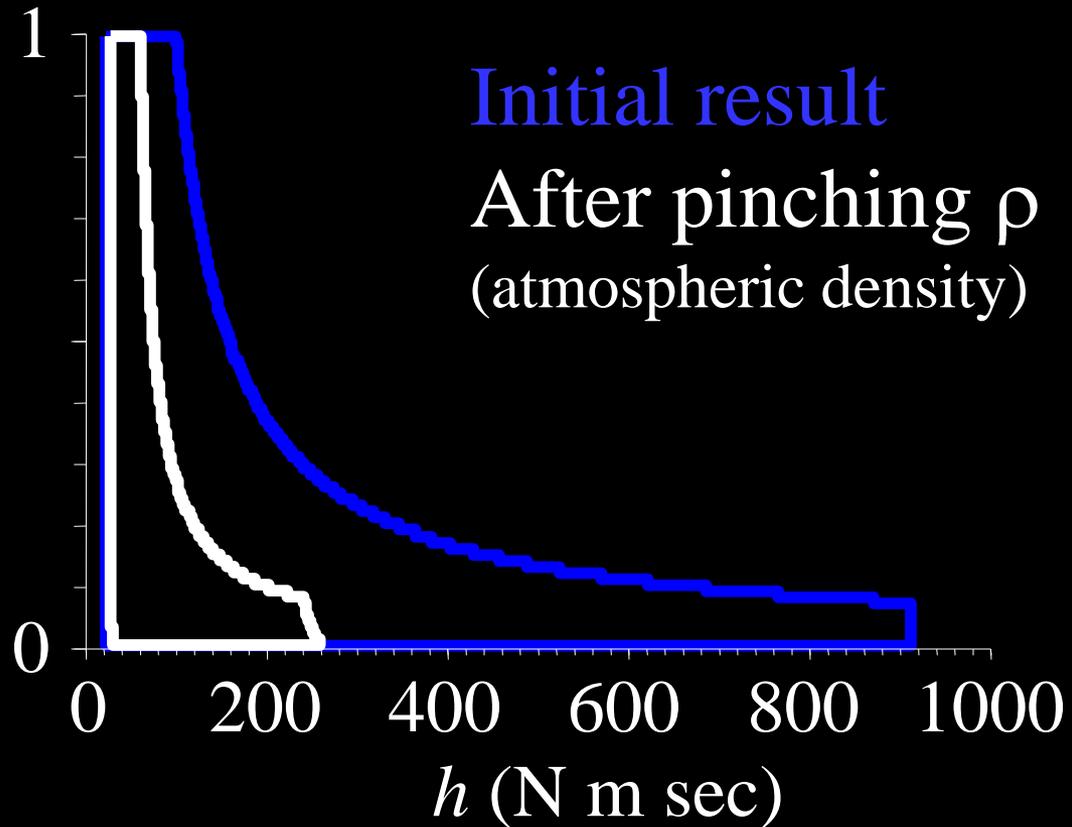
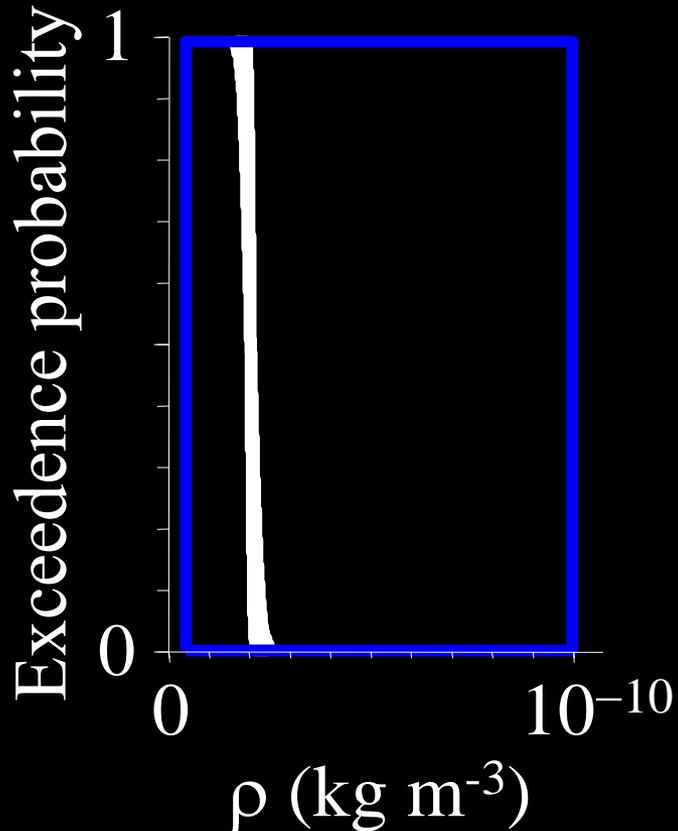


Required angular momentum h

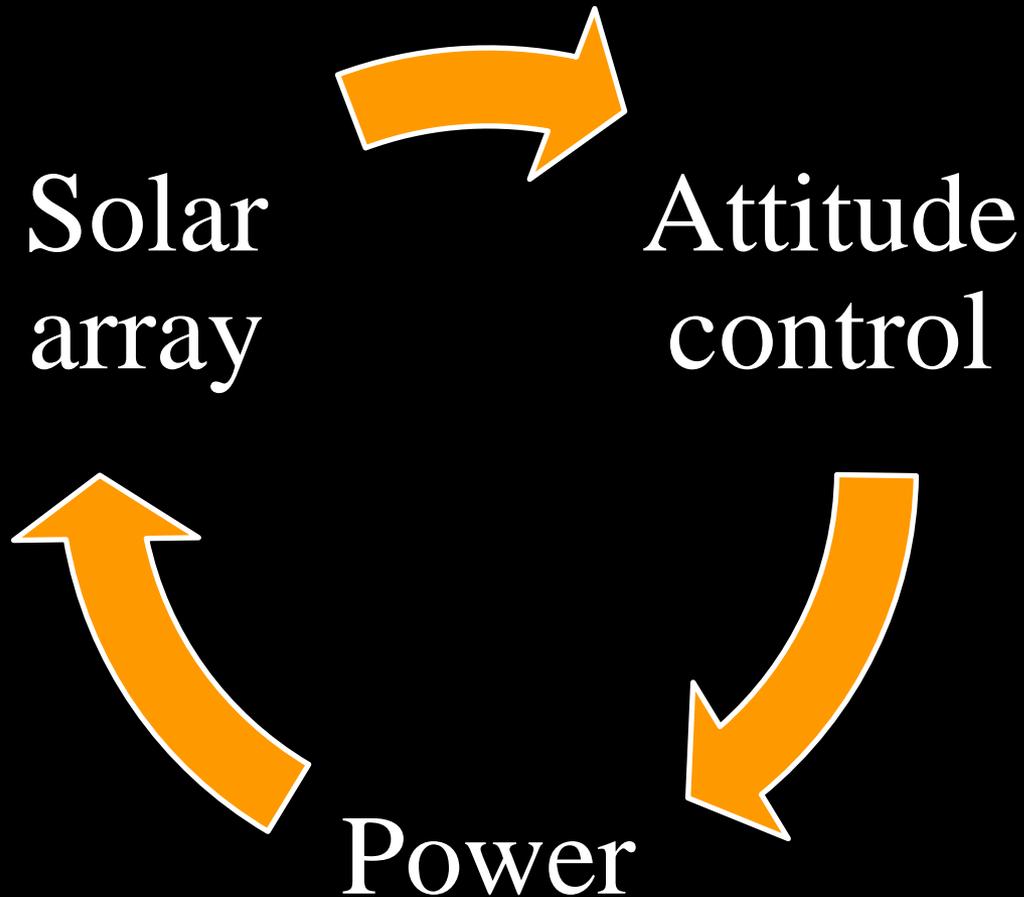


$$h = \Delta t_{orbit} \times \tau_{slew} + \tau_{dist}$$

Value of information: pinching ρ



Three linked subsystems



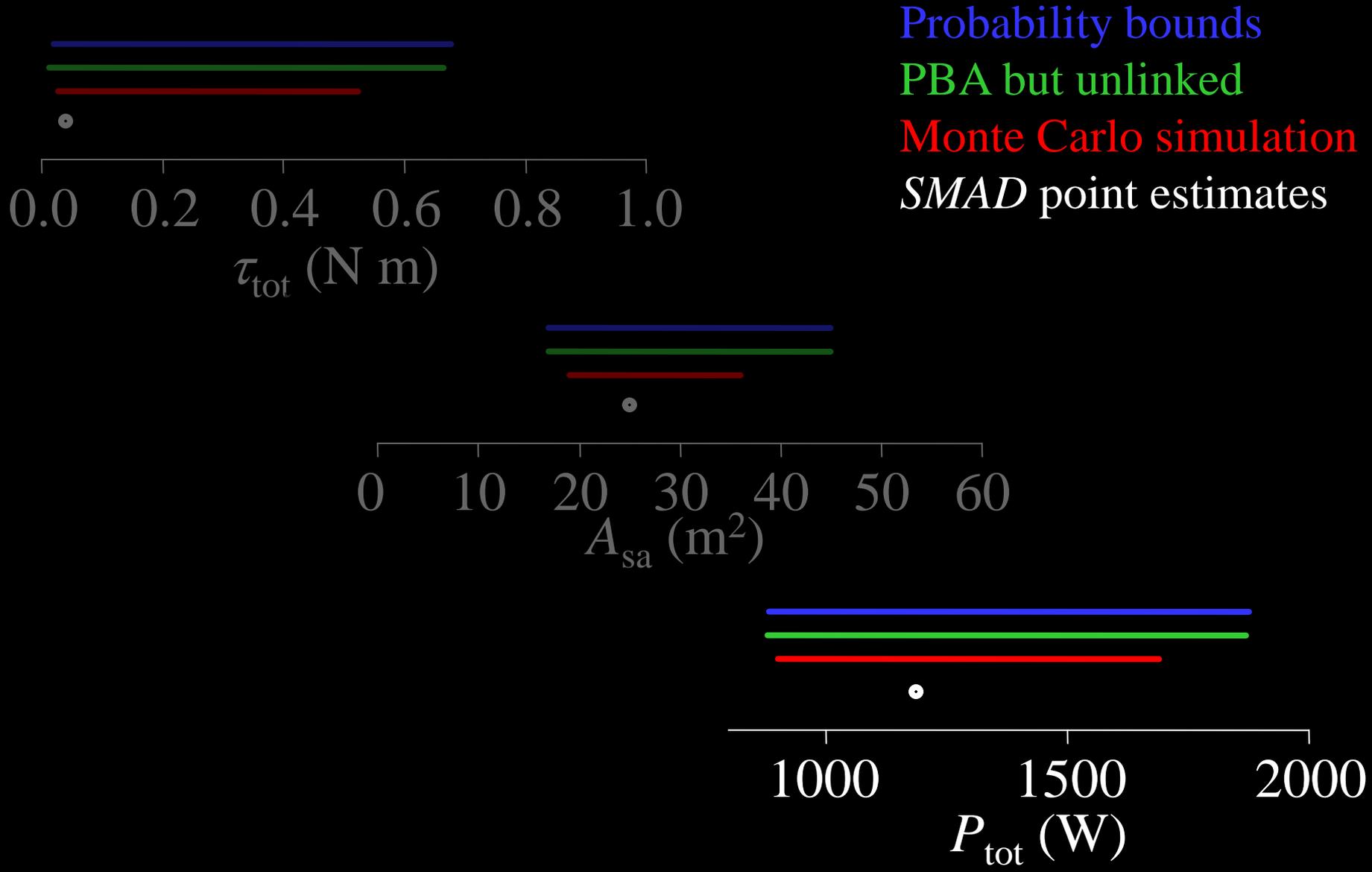
Variables passed iteratively

- Minimum moment of inertia I_{\min}
- Maximum moment of inertia I_{\max}
- Total torque τ_{tot}
- Total power P_{tot}
- Solar panel area A_{sa}

Analysis of calculations

- Need to check that original *SMAD* values and all Monte Carlo simulations are enclosed by p-boxes
- Need to ensure iteration through links doesn't cause runaway uncertainty growth (or reduction)
- Four parallel analyses
 - *SMAD*'s point estimates
 - Monte Carlo simulation
 - P-boxes but without linkage among subsystems
 - P-boxes with fully linked subsystems

Supports of results



Case study findings

- Different answers are consistent
 - Point estimates match the *SMAD* results
 - P-boxes span the points and the Monte Carlo intervals
- Calculations workable
 - No runaway inflation (or loss) of uncertainty
 - Easier than with Monte Carlo
- Practical and interesting results
 - Uncertainty can affect engineering decisions
 - Reducing uncertainty about ρ (by picking a launch date) strongly reduces design uncertainty

Accounting for epistemic and aleatory uncertainty in early system design

NASA SBIR Phase 2 Final Report

Applied Biomathematics
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Order Number: NNL07AA06C

July 2009

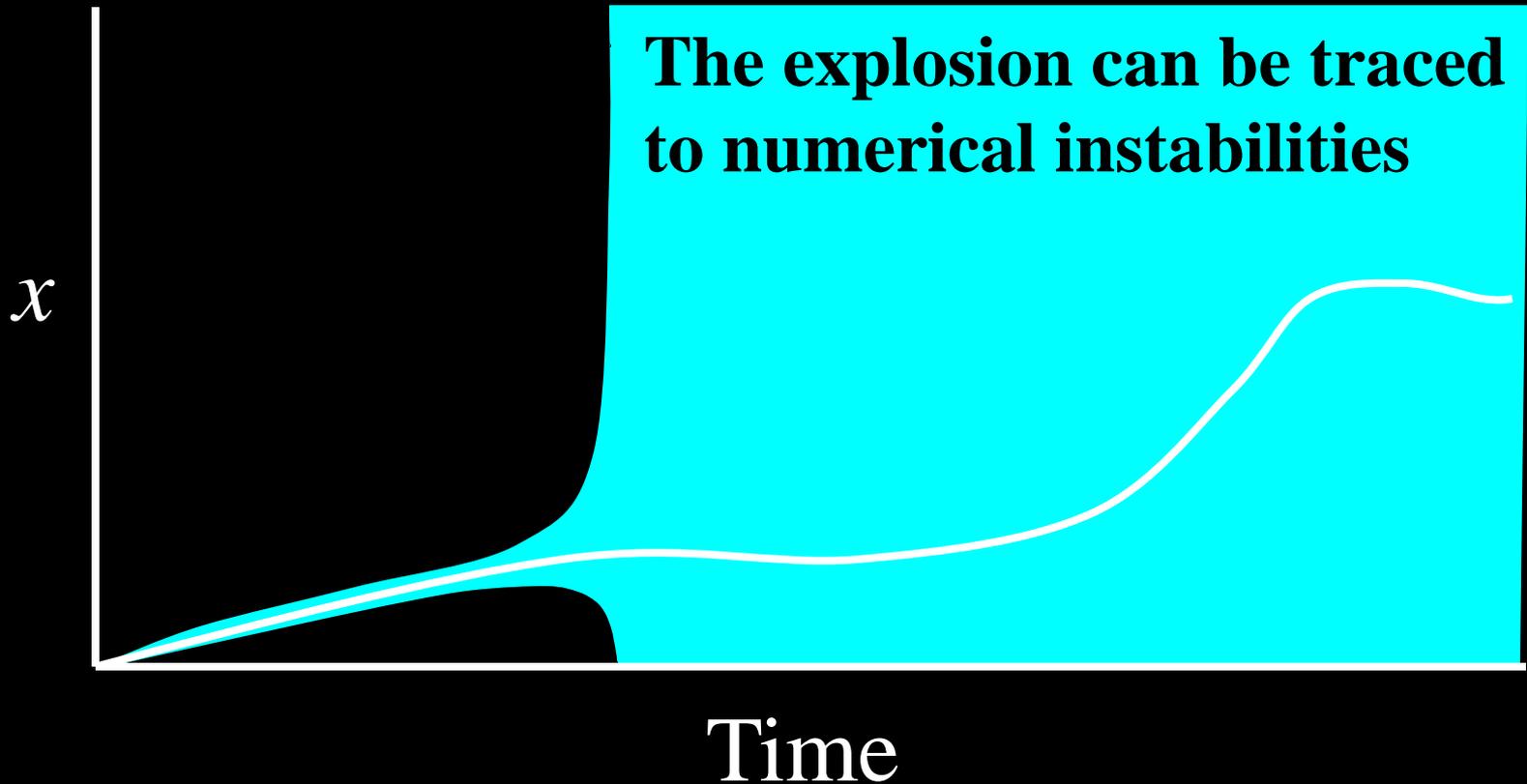
**For more information, consult
SBIR project report to NASA,
July 2009**

Uses for probability bounds analysis

- Uncertainty propagation
- Risk assessment
- Sensitivity analysis (for control and study)
- Reliability theory
- Engineering design ✓
- Validation
- Decision theory
- Regulatory compliance
- Finite element modeling
- Differential equations ✓

Differential equations

Uncertainty usually explodes



Uncertainty

- Artfactual uncertainty
 - Too few polynomial terms
 - Numerical instability
 - Can be reduced by a better analysis
- Authentic uncertainty
 - Genuine unpredictability due to input uncertainty
 - Cannot be reduced by a better analysis

Only by more information, data or assumptions

Uncertainty propagation

- We *want* the prediction to ‘break down’ if that’s what should happen
- But we don’t want artifactual uncertainty
 - Numerical instabilities
 - Wrapping effect
 - Dependence problem
 - Repeated parameters

Problem

- Nonlinear ordinary differential equation (ODE)

$$dx/dt = f(x, \theta)$$

with uncertain θ and uncertain initial state x_0

- Information about θ and x_0 comes as
 - Interval ranges
 - Probability distributions
 - Probability boxes

Model

Initial states (bounds)

Parameters (bounds)

Machina ex deo

V S P O D E

Mark Stadherr et al. (Notre Dame)

Taylor models

Interval Taylor series

**List of constants
plus remainder**

Example ODE

$$dx_1/dt = \theta_1 x_1(1 - x_2)$$

$$dx_2/dt = \theta_2 x_2(x_1 - 1)$$

What are the states at $t = 10$?

$$x_0 = (1.2, 1.1)^T$$

$$\theta_1 \in [2.99, 3.01]$$

$$\theta_2 \in [0.99, 1.01]$$

VSPODE

- Constant step size $h = 0.1$, Order of Taylor model $q = 5$,
- Order of interval Taylor series $k = 17$, QR factorization

VSPODE tells how to compute x_1

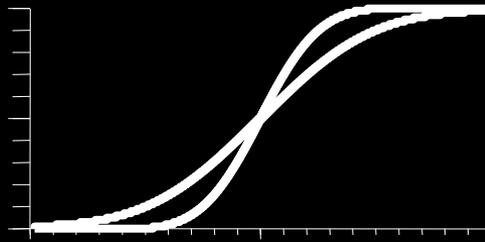
$$\begin{aligned} & 1.916037656181642 \times \theta_1^0 \times \theta_2^1 + 0.689979149231081 \times \theta_1^1 \times \theta_2^0 + \\ & -4.690741189299572 \times \theta_1^0 \times \theta_2^2 + -2.275734193378134 \times \theta_1^1 \times \theta_2^1 + \\ & -0.450416914564394 \times \theta_1^2 \times \theta_2^0 + -29.788252573360062 \times \theta_1^0 \times \theta_2^3 + \\ & -35.200757076497972 \times \theta_1^1 \times \theta_2^2 + -12.401600707197074 \times \theta_1^2 \times \theta_2^1 + \\ & -1.349694561113611 \times \theta_1^3 \times \theta_2^0 + 6.062509834147210 \times \theta_1^0 \times \theta_2^4 + \\ & -29.503128650484253 \times \theta_1^1 \times \theta_2^3 + -25.744336555602068 \times \theta_1^2 \times \theta_2^2 + \\ & -5.563350070358247 \times \theta_1^3 \times \theta_2^1 + -0.222000132892585 \times \theta_1^4 \times \theta_2^0 + \\ & 218.607042326120308 \times \theta_1^0 \times \theta_2^5 + 390.260443722081675 \times \theta_1^1 \times \theta_2^4 + \\ & 256.315067368131281 \times \theta_1^2 \times \theta_2^3 + 86.029720297509172 \times \theta_1^3 \times \theta_2^2 + \\ & 15.322357274648443 \times \theta_1^4 \times \theta_2^1 + 1.094676837431721 \times \theta_1^5 \times \theta_2^0 + \\ & [\underline{1.1477537620811058}, \underline{1.1477539164945061}] \end{aligned}$$

where θ 's are centered forms of the parameters; $\theta_1 = \theta_1 - 3$, $\theta_2 = \theta_2 - 1$

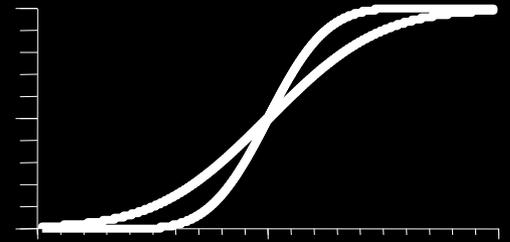
Input p-boxes

p-box

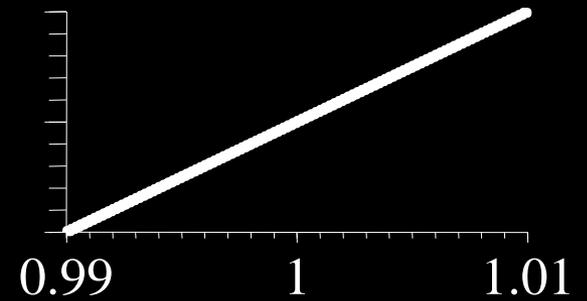
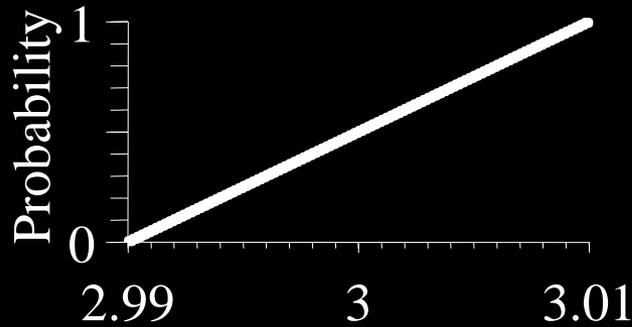
θ_1



θ_2

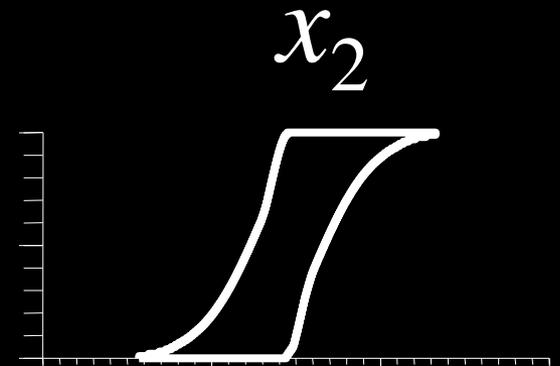
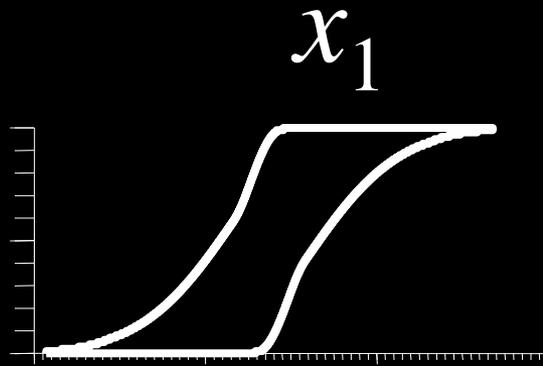


precise

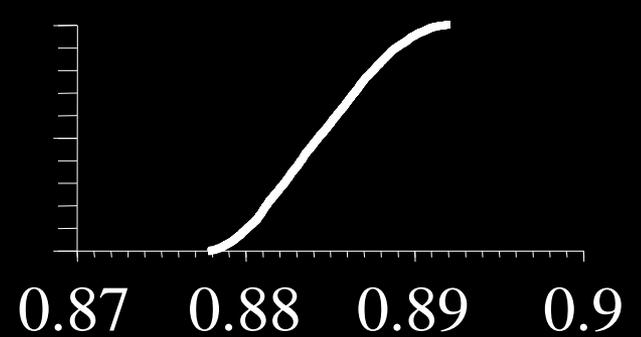
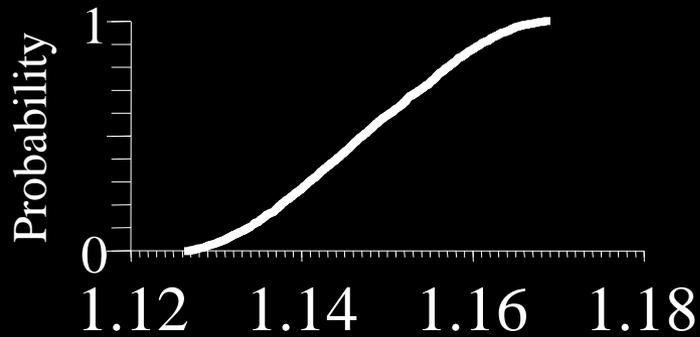


Results

p-box



precise



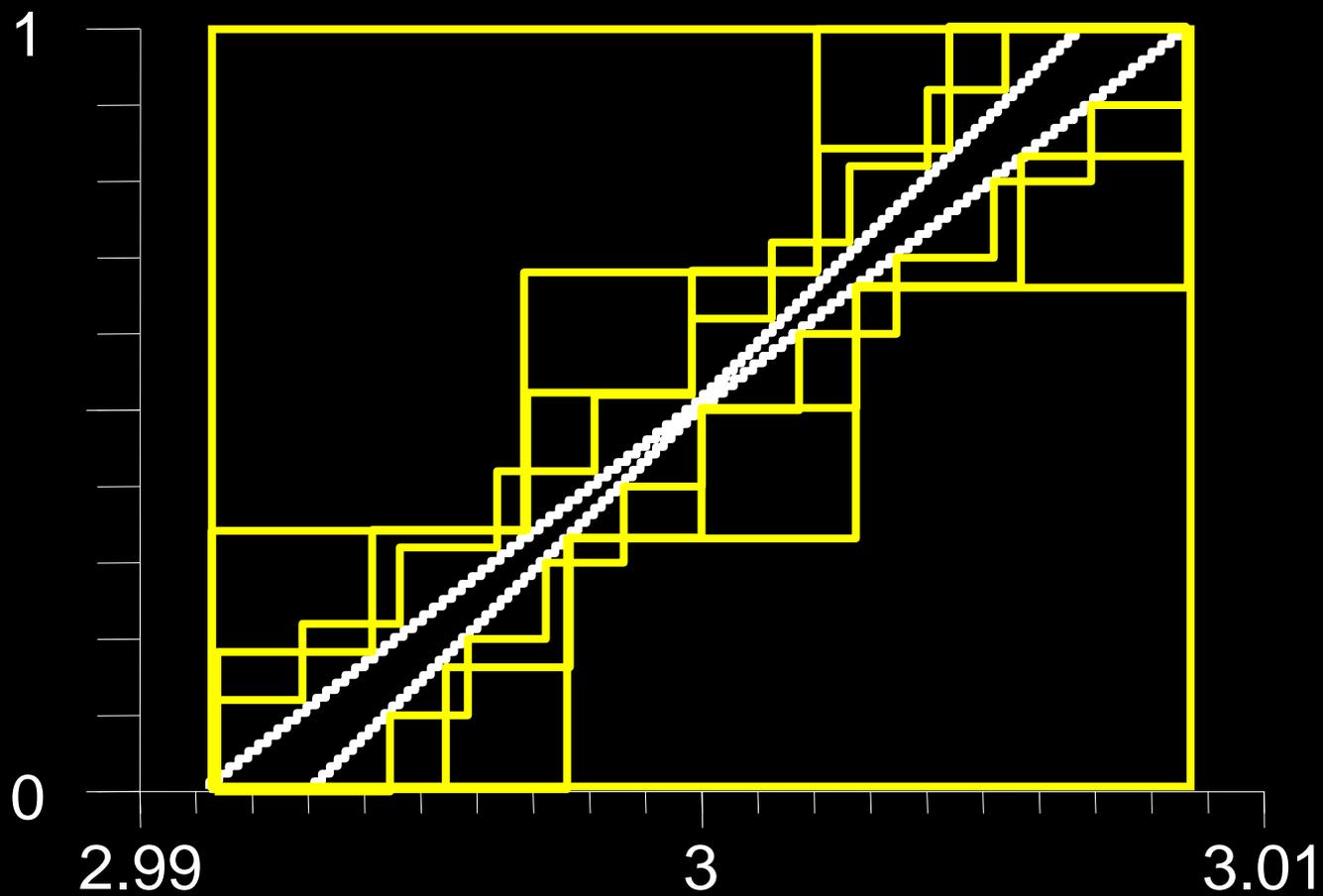
Still repeated uncertainties

$$\begin{aligned} & 1.916037656181642 \times \theta_1^0 \times \theta_2^1 + 0.689979149231081 \times \theta_1^1 \times \theta_2^0 + \\ & -4.690741189299572 \times \theta_1^0 \times \theta_2^2 + -2.275734193378134 \times \theta_1^1 \times \theta_2^1 + \\ & -0.450416914564394 \times \theta_1^2 \times \theta_2^0 + -29.788252573360062 \times \theta_1^0 \times \theta_2^3 + \\ & -35.200757076497972 \times \theta_1^1 \times \theta_2^2 + -12.401600707197074 \times \theta_1^2 \times \theta_2^1 + \\ & -1.349694561113611 \times \theta_1^3 \times \theta_2^0 + 6.062509834147210 \times \theta_1^0 \times \theta_2^4 + \\ & -29.503128650484253 \times \theta_1^1 \times \theta_2^3 + -25.744336555602068 \times \theta_1^2 \times \theta_2^2 + \\ & -5.563350070358247 \times \theta_1^3 \times \theta_2^1 + -0.222000132892585 \times \theta_1^4 \times \theta_2^0 + \\ & 218.607042326120308 \times \theta_1^0 \times \theta_2^5 + 390.260443722081675 \times \theta_1^1 \times \theta_2^4 + \\ & 256.315067368131281 \times \theta_1^2 \times \theta_2^3 + 86.029720297509172 \times \theta_1^3 \times \theta_2^2 + \\ & 15.322357274648443 \times \theta_1^4 \times \theta_2^1 + 1.094676837431721 \times \theta_1^5 \times \theta_2^0 + \\ & [1.1477537620811058, 1.1477539164945061] \end{aligned}$$

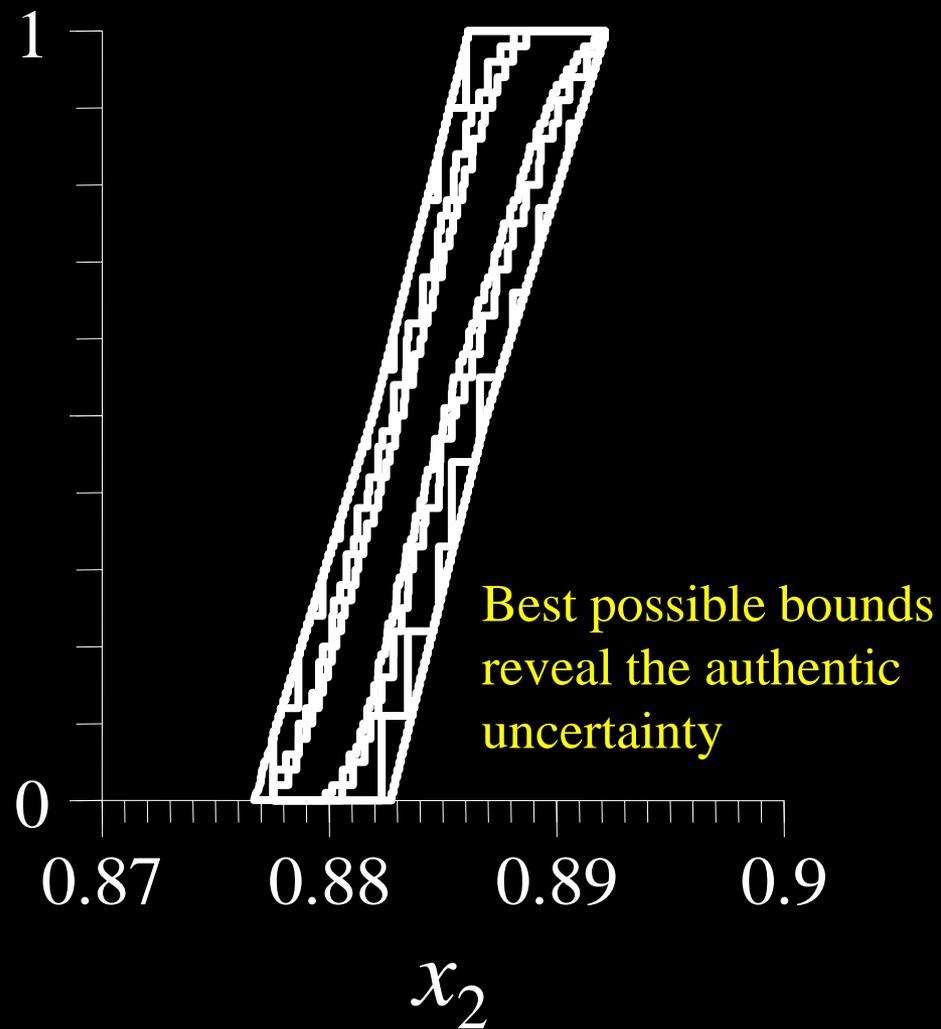
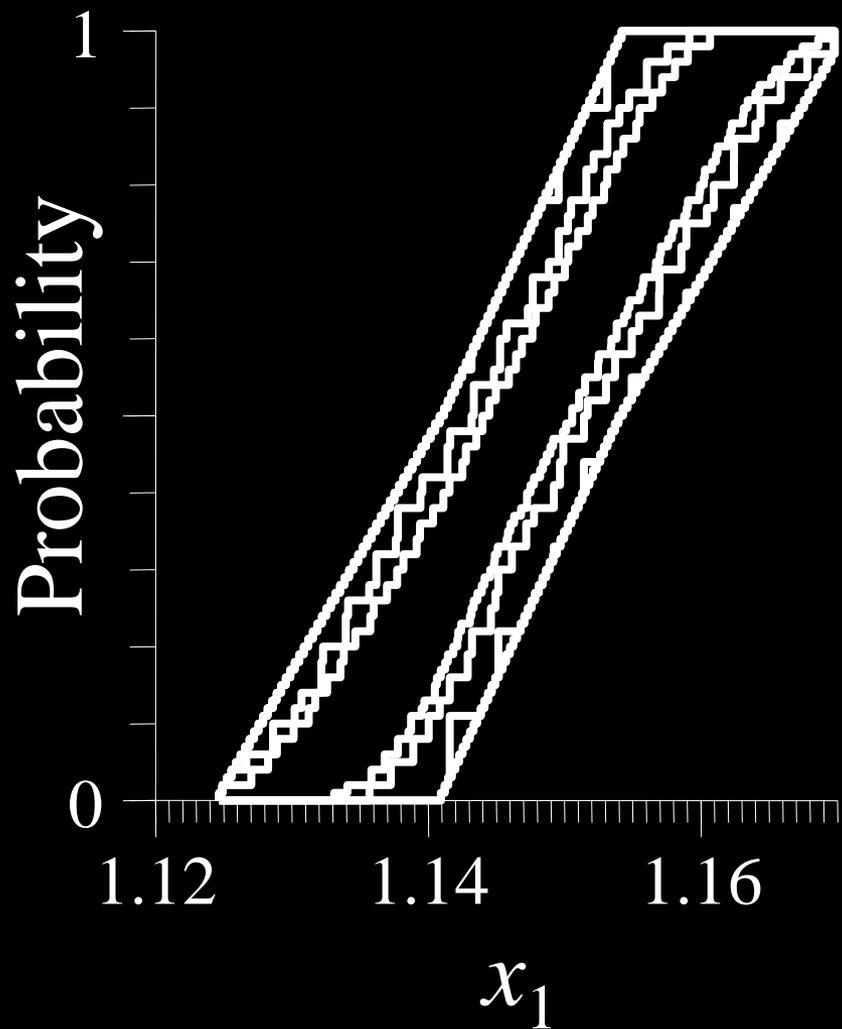
Subinterval reconstitution

- Subinterval reconstitution (SIR)
 - Partition the inputs into subintervals
 - Apply the function to each subinterval
 - Form the union of the results
- Still rigorous, but often tighter
 - The finer the partition, the tighter the union
 - Many strategies for partitioning
- Apply to *each cell* in the Cartesian product

Discretizations



Contraction from SIR



Monte Carlo is more limited

- Monte Carlo cannot propagate uncertainty
- Monte Carlo *cannot produce validated results*
 - Though can be checked by repeating simulation
- Validated results from distributions can be obtained by modeling inputs with (narrow) p-boxes and applying probability bounds analysis
- Results converge to narrow p-boxes obtained from infinitely many Monte Carlo replications

Results

- Probability bounds analysis with VSPODE are useful for bounding solutions of nonlinear ODEs

- They rigorously propagate uncertainty

about

{
Initial states
Parameters
}

in the form of

{
Intervals
Distributions
P-boxes
}

Probability Bounds Analysis for Nonlinear Dynamic Process Models

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Dynamic process models frequently involve uncertain parameters and inputs. Propagating these uncertainties rigorously through a mathematical model to determine their effect on system states and outputs is a challenging problem. In this work, we describe a new approach, based on the use of Taylor model methods, for the rigorous propagation of uncertainties through nonlinear systems of ordinary differential equations (ODEs). We concentrate on uncertainties whose distribution is not known precisely, but can be bounded by a probability box (p-box), and show how to use p-boxes in the context of Taylor models. This allows us to obtain p-box representations of the uncertainties in the state variable outputs of a nonlinear ODE model. Examples having two to three uncertain parameters or initial states and focused on reaction process dynamics are used to demonstrate the potential of this approach. Using this method, rigorous probability bounds can be determined at a computational cost that is significantly less than that required by Monte Carlo analysis. © 2010 American Institute of Chemical Engineers AIChE J, 57: 404–422, 2011

Keywords: design (process simulation), mathematical modeling, numerical solutions, reactor analysis, bioprocess engineering

Introduction

Systems of ordinary differential equations (ODEs) are the basis for many mathematical models in engineering and science. For example, models of reactor dynamics are based on unsteady-state material and energy balances and thus take

the form of a system of first-order ODEs, which typically is nonlinear. Generally, the problem of interest is an initial value problem (IVP), in which an initial state is given and the system then integrated numerically until some final time (time horizon) is reached, thus determining numerical approximations of the final state, as well as of the trajectory followed to reach it.

Often these dynamic models involve uncertainties in parameters and/or initial states. Analysis of the impact of such uncertainties is clearly important in models of process dynamics, as used, for example, in state and parameter estimation^{1,2}

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Current Address of Youdong Lin: LINDO Systems, Inc., 1615 North Dayton St., Chicago, IL 60642.

PBA relaxes assumptions

- Everyone makes assumptions, but not all sets of assumptions are equal:

Linear

Normal

Independence

Monotonic

Unimodal

Known correlation

Any function

Any distribution

Any dependence

- PBA doesn't require unwarranted assumptions

Wishful thinking

Analysts often make convenient assumptions that are not really justified:

1. Variables are independent of one another
2. Uniform distributions model gross uncertainty
3. Distributions are stationary (unchanging)
4. Distributions are perfectly precisely specified
5. Measurement uncertainty is negligible

You don't have to think wishfully

A p-box can discharge a false assumption:

1. Don't have to assume any dependence at all
2. An interval can be a better model of uncertainty
3. P-boxes can enclose non-stationary distributions
4. Can handle imprecise specifications
5. Measurement data with plus-minus, censoring

Rigorousness

- “Automatically verified calculations”
- The computations are guaranteed to enclose the true results (so long as the inputs do)
- You can still be wrong, but the *method* won't be the reason if you are

Take-home messages

- Using bounding, you don't have to pretend you know a lot to get quantitative results
- Probability bounds analysis bridges worst case and probabilistic analyses in a way that's faithful to both and makes it suitable for use in early design

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Chris Paredis, Georgia Tech

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Electric Power Research Institute (EPRI)

Sandia National Laboratories

End